

# Multi-Scale, Multi-Fidelity Systems Design and Simulation Environment

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# Technology Motivation

- DoD systems are increasingly complex and challenge human cognitive, and organizational, abilities.
  - Model fidelity and connecting multi-fidelity models to coherent system views.
    - Using the lowest, most appropriate level of fidelity.
  - Engineering model robustness.
  - Reuse of models.
  - Handling highly scaled simulation problems – “digital twin”.
  - Discovering unforeseen behavior.
  - Understanding complex results.
  - Accurate simulations, well before we commit.
  - Simulations & models that live with us throughout the program’s life.

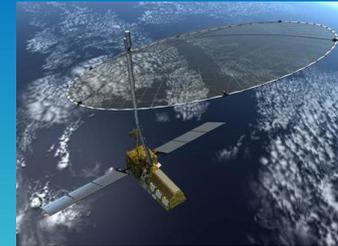


# The Basics

**Simulating the System**

Simulation

SystemVerilog



**Creating a system from subsystems**

System

Satellite

**Making subsystems from models**

Subsystem

Orbit

Subsystem

Battery

Subsystem

Solar Panel

Model

Universal

Model

Elliptical

Model

Circular

Model

1-D

Model

Hi Fidelity

Model

Lo Fidelity

**Building models from 1<sup>st</sup> principles**

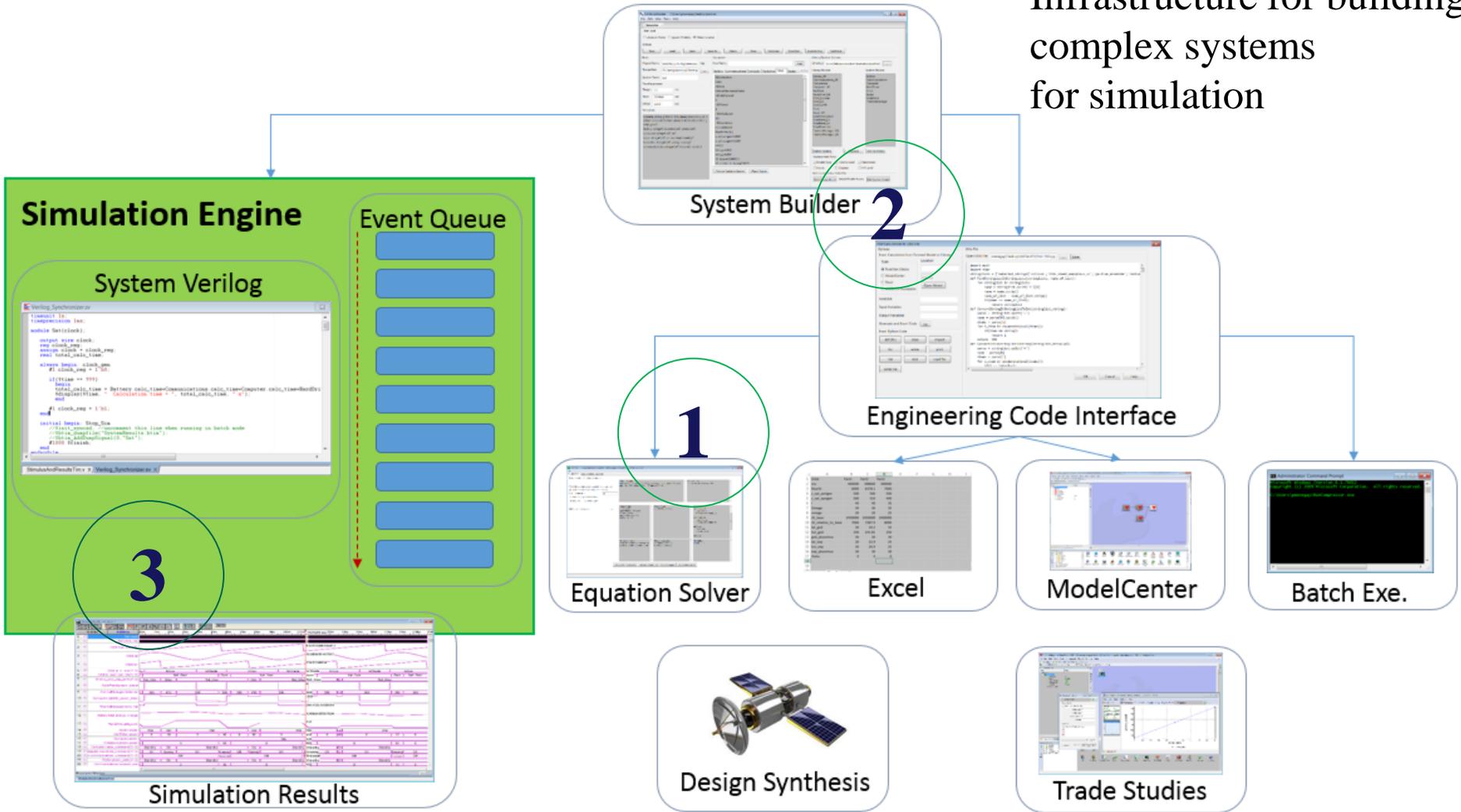
# Some Observations

- Most simulation systems are not scalable. But those in the EDA space are. Why not use a paradigm like SystemVerilog to achieve scale?
- Most model-building tools are not generic, and if they are, are not robust enough to be used within simulation. We need to improve the areas where generic numerical methods are weak.
- Most system building tools have trouble handling fidelity transitions as models get more complex. Tools also tend to be domain specific. This leads to bifurcation of modeling efforts.
- System simulation needs to be able to *drive* other tools & models. It also needs to be *drivable* from other tools & models.
- When results from large scale simulations come in, they are often hard to interpret. How can we create views that make sense?



# System Computation Platform

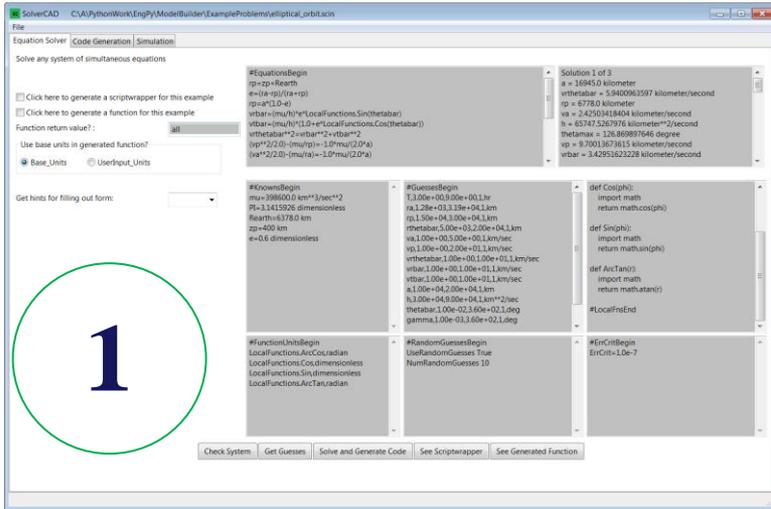
Infrastructure for building complex systems for simulation



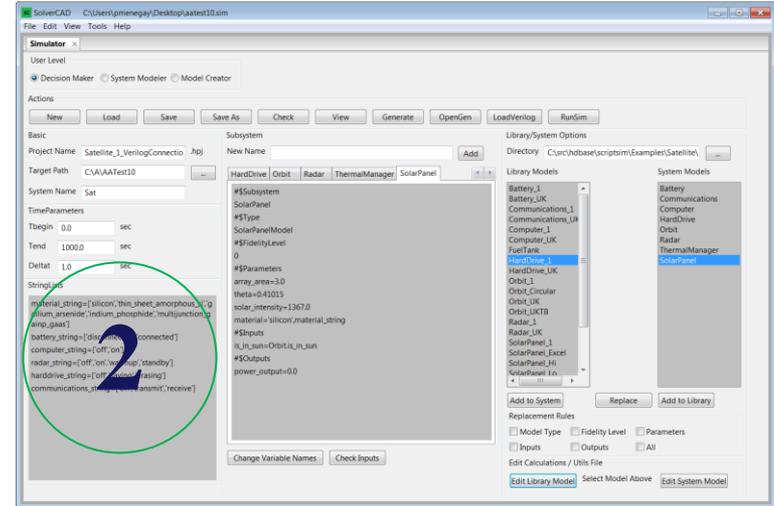
Our approach is to fix weaknesses in the 3 key areas



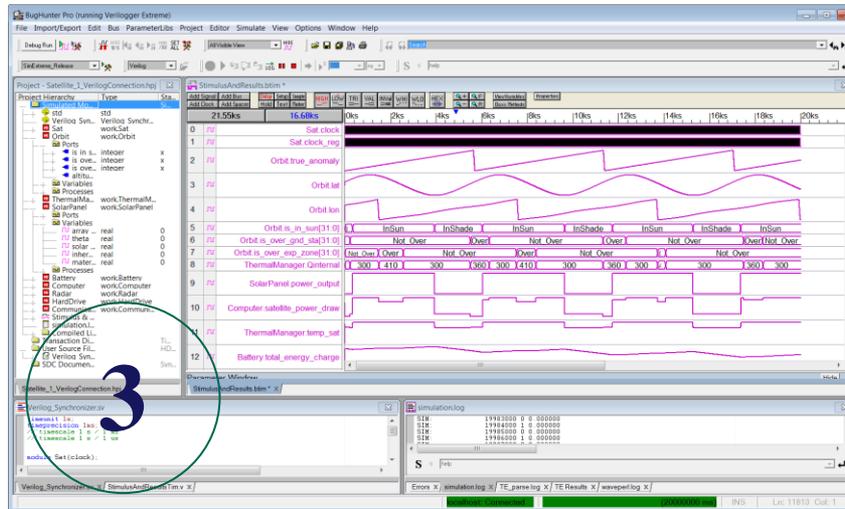
# 3 Core Elements & Workflow



**Eqn. Based Model Creator** -- Simultaneous equation solver for creating new engineering models for subsystems.



**System Builder** -- User interface for creating new system models.



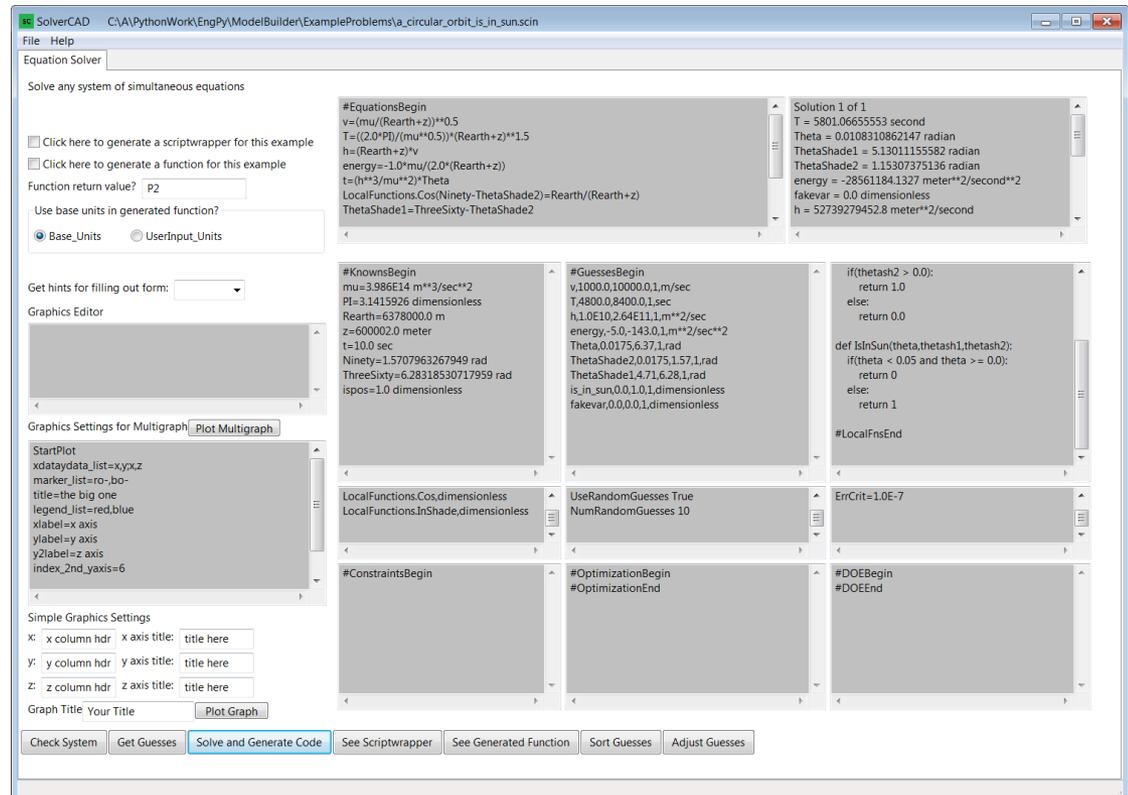
**System Simulator** -- SystemVerilog simulation engine for system model execution and results.



# 1. Eqn. Based Model Creator

A general purpose model-creation environment for engineering analysis. Under development.

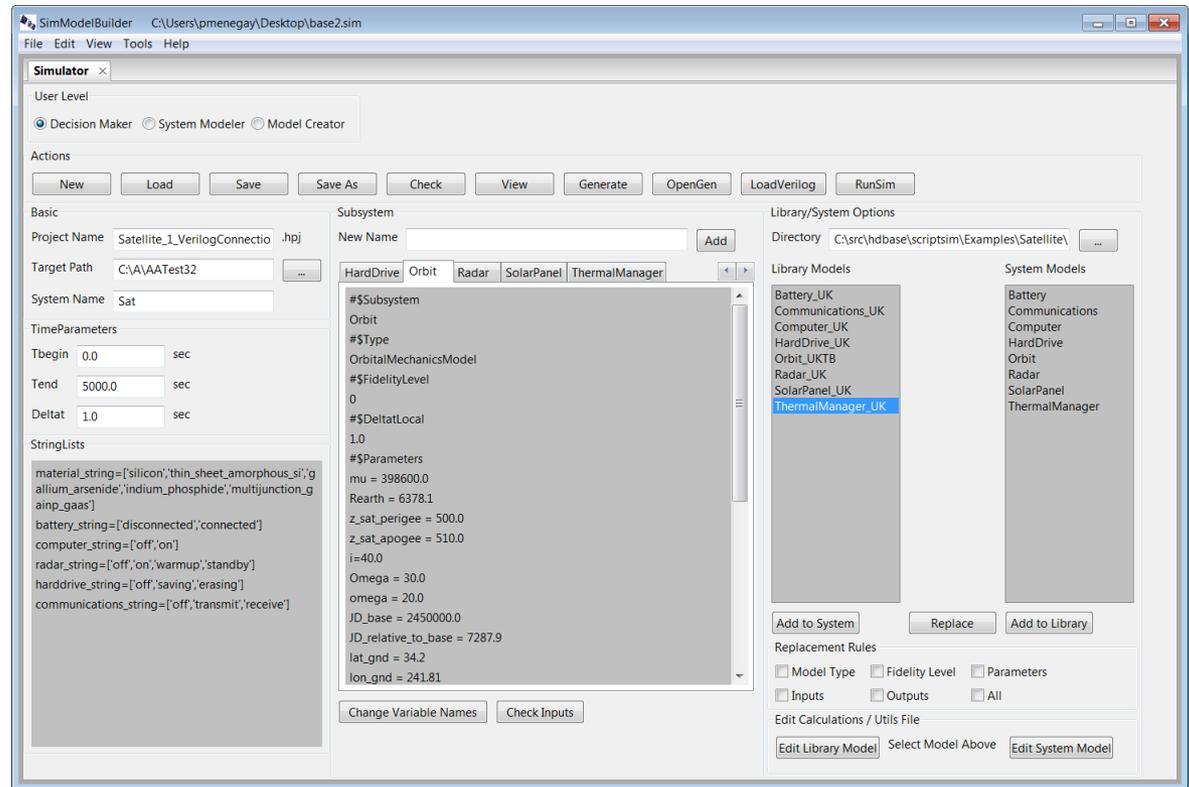
- Solves any system of nonlinear simultaneous equations.
- Manages the core numerical library to achieve robustness.
- Generates Python functions to use in simulation code.
- Uses a library concept for storing functions for later use.



## 2. System Builder

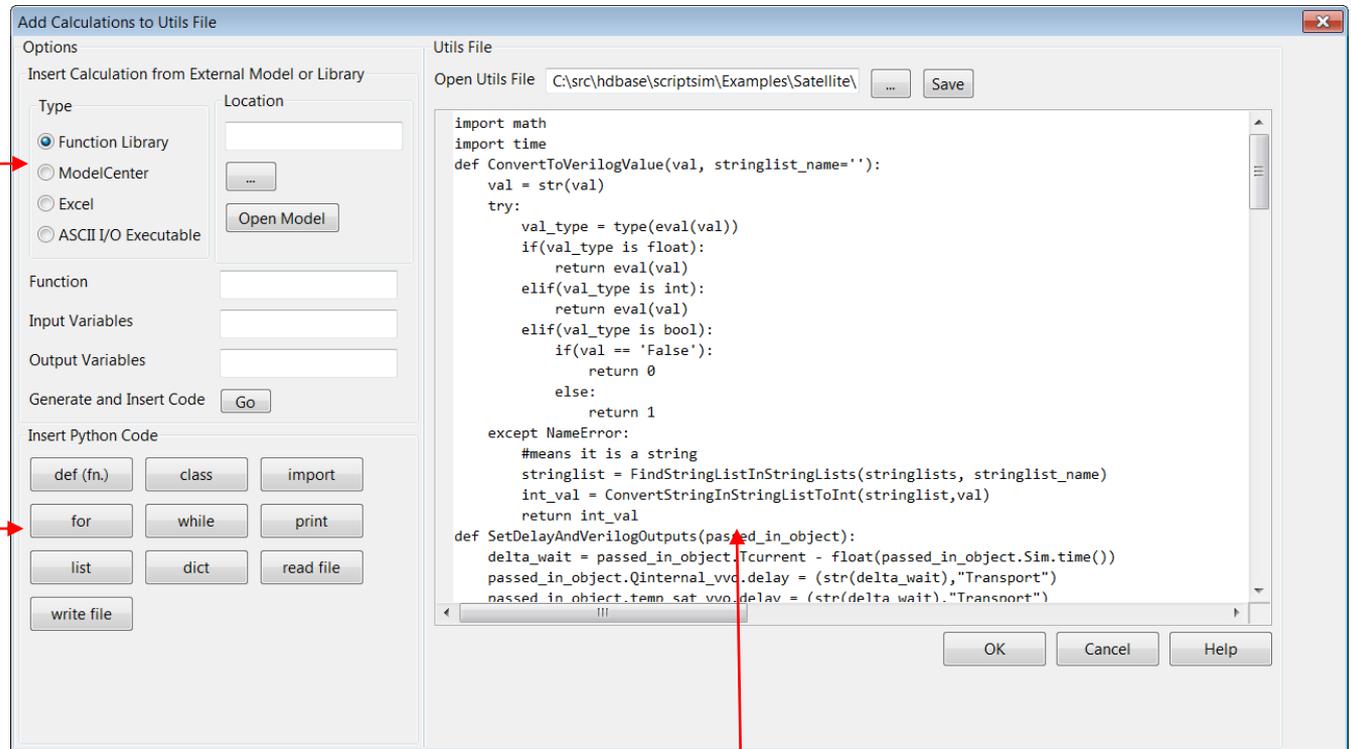
A GUI that helps you build, connect, and modify sub-system models for simulation.

- Create system models from a library of pre-built subsystems.
- Create subsystem models from equation solver and external tools.
- Publish subsystems to library.
- Easily replace subsystems with higher/lower fidelity ones.



## 2. System Builder, cont.

Includes a feature to help you edit calculations or link to them from external tools.



Link to pre-built models in SolverCAD, ModelCenter, Excel, or your own program.

Buttons for quickly creating common Python constructs.

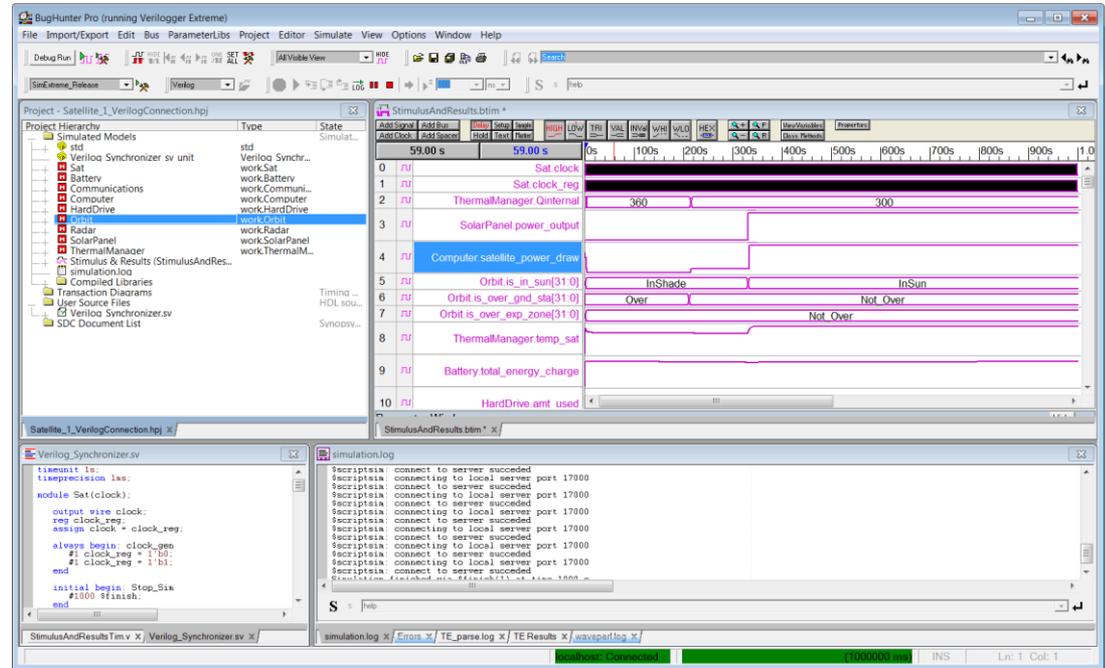
Edit the calculation file for any subsystem in Python.



# 3. System Simulator

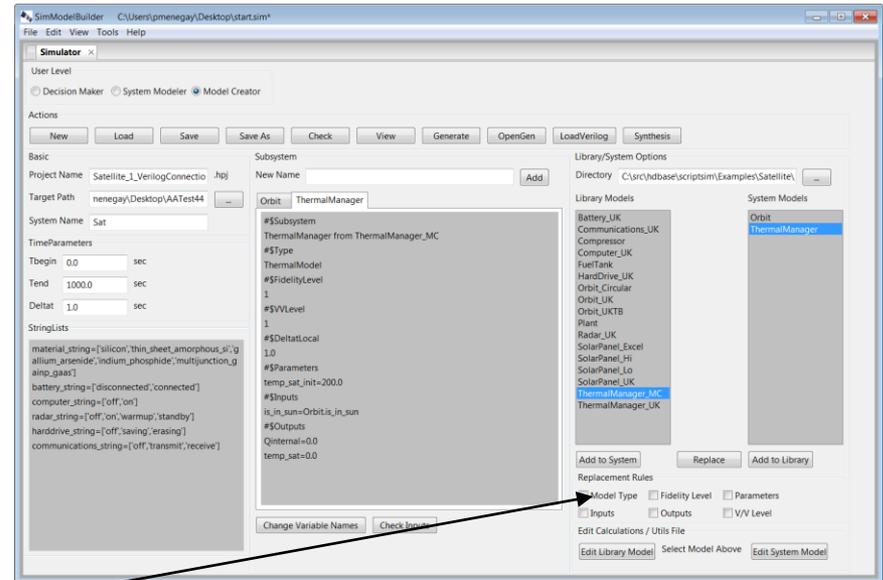
The system simulator combines a high-performance compiled-code SystemVerilog simulator with a Python interpreter to enable engineering level modeling of real world systems.

- Timing diagram with simulation results
- Generated SystemVerilog code.
- Hierarchical view of subsystems and components.
- Full IDE including single step debugging, breakpoints, etc.
- Design browsing & navigation.
- Various output formats



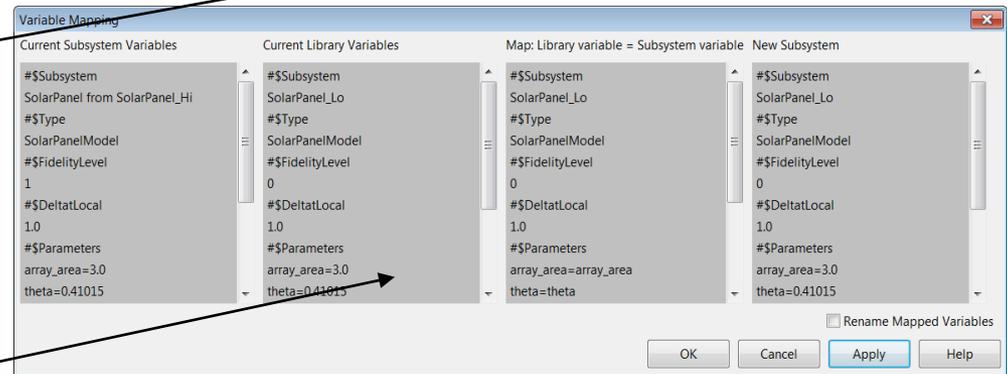
# Multifidelity Modeling

- Models of different fidelity can be switched on the fly.
- As the project advances, the simulation environment remains in place, and maintains connectivity with previous models.



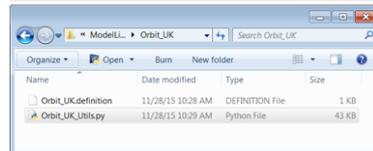
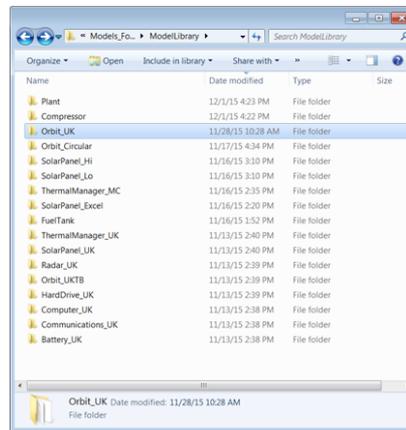
**Replacement rules for switching model fidelity**

**Variable mapping to ensure continuity between models**



# Model Libraries

- Subsystem model library.
  - Orbit calculations, solar panel, battery, etc. are publishable and retrievable from library.
- Function library for equation solver.
  - Generated functions can be accessed by System Builder.

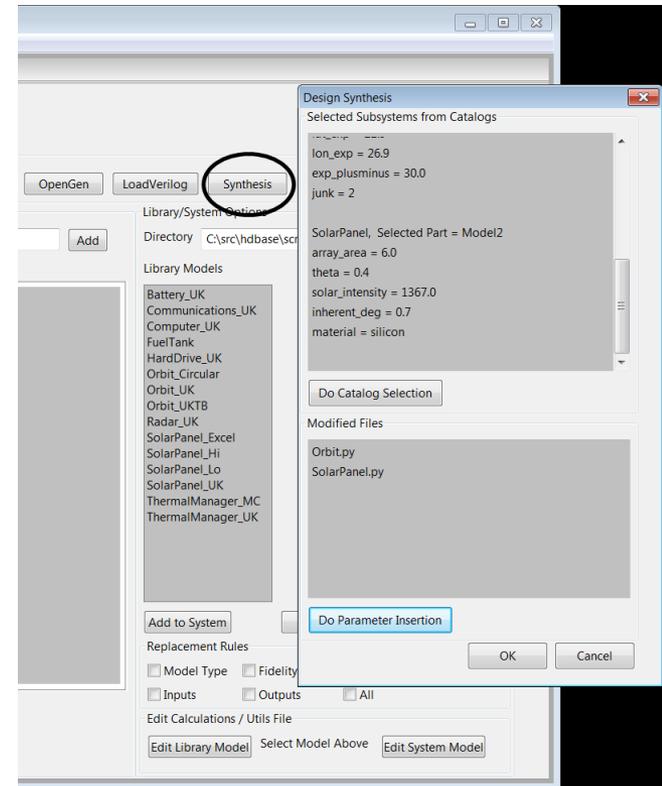


```
Orbit_UK_definition
1 #Subsystem
2 orbit_UK
3 #type
4 OrbitalMechanicsModel
5 #fidelitylevel
6 0
7 #delta_tlocal
8 1.0
9 #Parameters
10 mu = 398600.0
11 Roarth = 6378.1
12 z_sat_perigee = 500.0
13 z_sat_apogee = 510.0
14 i=40.0
15 Omega = 30.0
16 omega = 20.0
17 J0_base = 2450000.0
18 J0_relative_to_base = 7287.9
19 lat_gnd = 34.2
20 lon_gnd = 241.81
21 gnd_plusminus = 30.0
22 lat_exp = 22.9
23 lon_exp = 26.9
24 exp_plusminus = 30.0
25 theta = 0.0
26 #Inputs
27 None
28 #Outputs
29 true_anomaly = 0.0
30 lat = 0.0
31 lon = 0.0
32 is_in_sun=False
33 is_over_gnd_sta=False
34 is_over_exp_zone=False
35 altitude = 505.0
length Ln:15 Col:15 Sel:0|0 Dos:Windows UTF-8 INS
```



# Synthesis

- Compares simulation subsystems with a catalog of parts.
  - User has presumably optimized the subsystem and now wants to select hardware.
  - Software will choose the closest part from catalog and resimulate.



# Satellite Model

Satellite circles the earth in a standard elliptical orbit. It's mission is to collect earth data over an experimental zone and download it to a ground station at another location. It charges a battery in the sun and depletes the battery in the shade. The simulation objective is to understand if the subsystems are sized properly.

#####_Orbit_#####
PAR mu=398600.0
PAR Rearth=6378.1
PAR z_sat_perigee=500.0
PAR z_sat_apogee=510.0
PAR i=40.0
PAR Omega=30.0
PAR omega=20.0
PAR JD_base=2450000.0
PAR JD_relative_to_base=7287.9
PAR lat_gnd=34.2
PAR lon_gnd=241.81
PAR gnd_plusminus=30.0
PAR lat_exp=22.9
PAR lon_exp=26.9
PAR exp_plusminus=30.0
PAR theta=0.0
INP None
OUT true_anomaly=0.0
OUT lat=0.0
OUT lon=0.0
OUT is_in_sun=False
OUT is_over_gnd_sta=False
OUT is_over_exp_zone=False
OUT altitude=505.0

#####_SolarPanel_#####
PAR array_area=3.0
PAR theta=0.41015
PAR solar_intensity=1367.0
PAR inherent_deg=0.77
PAR material='silicon',material_string
INP is_in_sun=Orbit.is_in_sun
OUT power_output=0.0

Sat
tbegin=0.0
tend=1000.0
deltat=100.0

#####_Computer_#####
PAR power_rating=100.0
PAR on_state_init='on',computer_string
INP solar_panel_power_output=SolarPanel.power_output
INP radar_power=Radars.power
INP harddrive_power=HardDrive.power
INP comm_power=Communications.power
INP is_over_gnd_sta=Orbit.is_over_gnd_sta
INP is_over_exp_zone=Orbit.is_over_exp_zone
OUT satellite_power_draw=0.0
OUT on_state='off',computer_string
OUT power=0.0
OUT radar_command='off',radar_string
OUT harddrive_command='off',harddrive_string
OUT communications_command='off',communications_string

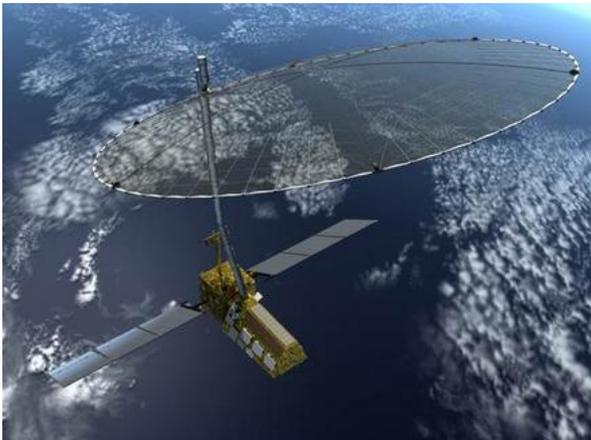
#####_Battery_#####
PAR capacity_rating=360000.0
PAR voltage_rating=12.0
PAR total_energy_charge_init=2160000.0
PAR con_state='connected',battery_string
INP satellite_power_draw=Computer.satellite_power_draw
OUT total_energy_charge=0.0

#####_Radar_#####
PAR power_on=300.0
PAR power_standby=200.0
PAR power_warmup=100.0
PAR power_off=0.0
INP command=Computer.radar_command
OUT power_state='off',radar_string
OUT power=0.0

#####_HardDrive_#####
PAR storage_capacity=650.0
PAR store_speed=0.5
PAR erase_speed=1.0
PAR power_rating=10.0
PAR amt_used_init=300.0
INP command=Computer.harddrive_command
OUT on_state='off',harddrive_string
OUT store_rate=0.0
OUT erase_rate=0.0
OUT power=0.0
OUT amt_used=0.0

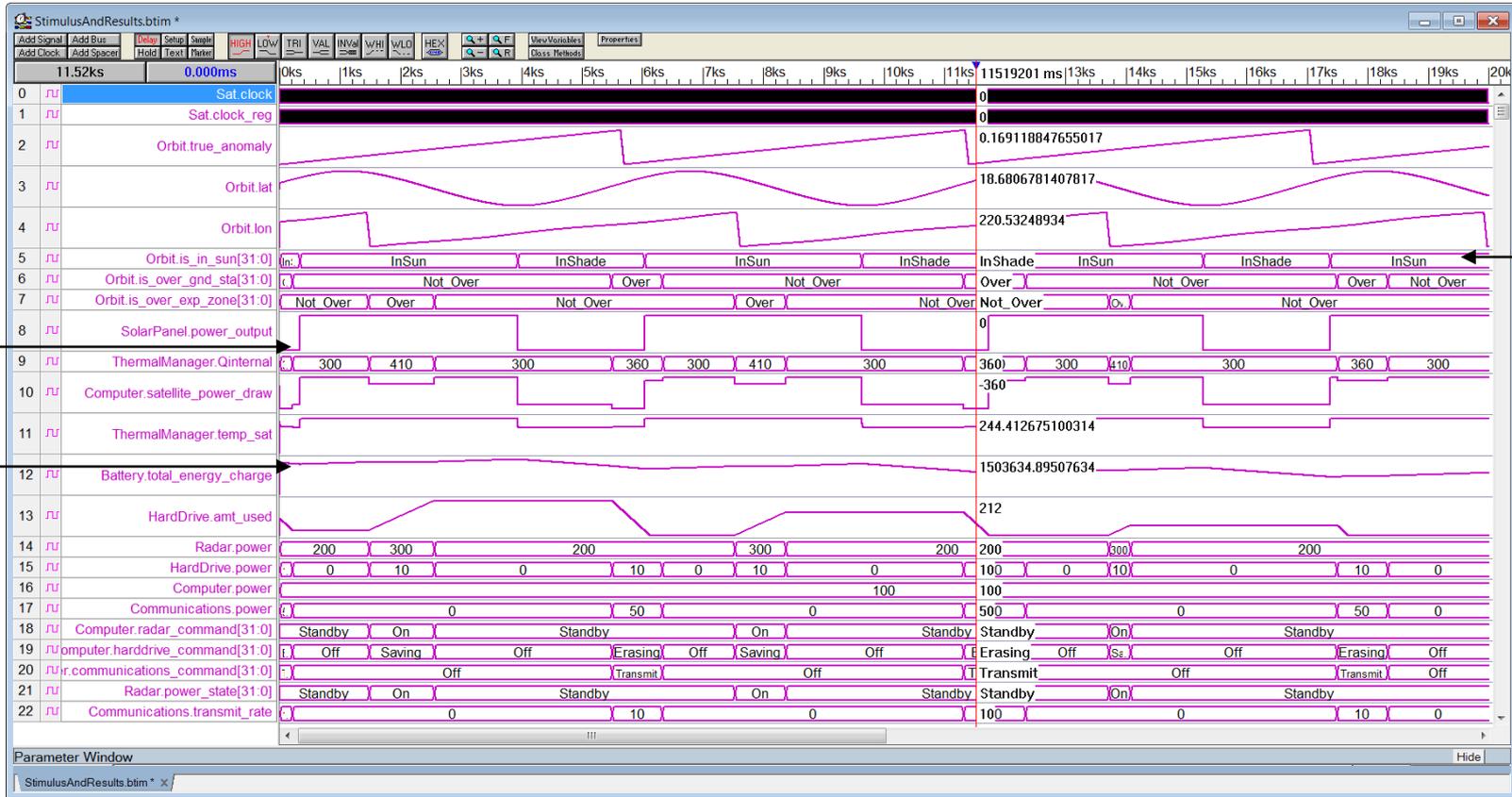
#####_Communications_#####
PAR transmit_speed=10.0
PAR receive_speed=20.0
PAR power_rating=50.0
INP command=Computer.communications_command
OUT on_state='off',communications_string
OUT transmit_rate=0.0
OUT receive_rate=0.0
OUT power=0.0

#####_ThermalManager_#####
PAR earth_albedo=0.3
PAR solar_intensity=1388.0
PAR stefan_boltzmann_const=5.67e-8
PAR temp_earth=290.0
PAR temp_sun=5800.0
PAR temp_space=2.7
PAR radius_earth=6378.1*1000.0
PAR sat_diameter=1.0
PAR sat_absorptivity=0.7
PAR sat_emissivity=0.9
PAR earth_emissivity=0.5
PAR sat_thermal_capacity=50.0
PAR temp_sat_init=300.0
PAR sat_weight=10.0
INP is_in_sun=Orbit.is_in_sun
INP sat_altitude=Orbit.altitude
INP radar_power=Radars.power
INP computer_power=Computer.power
INP harddrive_power=HardDrive.power
INP comm_power=Communications.power
OUT Qinternal=0.0
OUT temp_sat=0.0



NASA / JPL

# Results



**Battery slowly drains to 0**

**Solar Panel does not recharge it when exposed to sun  
I.e, the Solar Panel is undersized. Battery is oversized.**

# Results, cont.

- One way to vary the solar panel / battery size is to use constrained randomization.
- Solution was to increase the solar panel area from 3.0 to 4.0 m\*\*2 and decrease the battery capacity from 360,000 to 60,000 amp-sec.

```
Constrain_Batt_SolarPanel.vh
program Constrain_Batt_SolarPanel;
class Constrainer;
rand integer cr_to_aar;
rand integer cr;
rand integer aar;

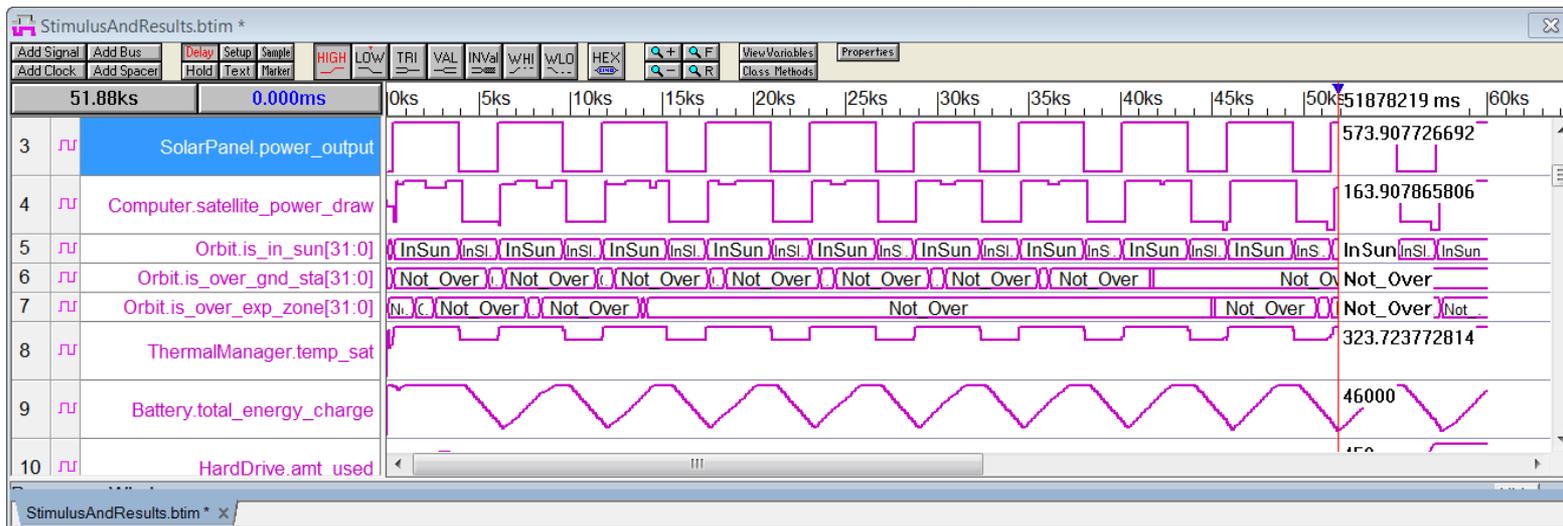
constraint c1 { cr_to_aar < 20000; }
constraint c2 { cr_to_aar > 10000; }

constraint c3 { cr >= 55000; }
constraint c4 { cr <= 200000; }

constraint c5 { aar >= 4; }
constraint c6 { aar <= 10; }

constraint c0 { cr_to_aar == cr / aar; }
endclass

initial
begin
Constrainer obj = new();
int wfile;
wfile = $fopen("c:/users/paenegay/desktop/dae.csv", "w");
$write(wfile, "capacity_rating, array_area \n");
for(int i=0; i<100; i++)
if(obj.randomize())
begin
$write(wfile, "%0d , %0d \n", obj.cr, obj.aar);
$display(" Randomization successful : cr = %0d aar = %0d cr_to_aar = %0d", obj.cr, obj.aar, obj.cr_to_aar);
end
else
$display("Randomization failed");
$display(obj.cr_to_aar);
end
//fclose(wfile);
endprogram
```



# Results, cont.

- This could also have been achieved by driving the simulator from a ModelCenter DOE.

The screenshot displays the Phoenix Integration ModelCenter 11.2 interface. The main window shows a Component Tree with variables: cr\_to\_aar (0), capacity\_rating (0), array\_area (0), and cr\_to\_aar (0). The DOE Tool window is open, showing a Design Table for 'cr\_to\_aar2' with 100 runs. The Data Explorer (DOE) window is also open, displaying a table of results for the design variable 'Model.cr\_to\_aar.capacity\_rating' and the response 'Model.cr\_to\_aar.cr\_to\_aar'.

Legend:	input	valid output	invalid output	modified value			
		95	96	97	98	99	100
design variable [Model.cr_to_aar.capacity_rating]	171188	183982	184004	186930	196412	199706	
design variable [Model.cr_to_aar.array_area]	10	10	10	10	10	10	
response [Model.cr_to_aar.cr_to_aar]	17118.8	18398.2	18400.4	18693	19641.2	19970.6	

# Overall Results

- Once engineering models were made, system integration was fast, 1-2 days for this case.
  - Model libraries were key.
- Provision for multi-fidelity model switching allowed project to remain within a single environment throughout its life.
- Scalability tests on a simple vehicle object lends credence to the SystemVerilog approach.
  - SystemVerilog can simulate up to memory limits of computer. 18 million vehicles for 32-bit and 40 million for 64 bit.
  - SimPy by contrast could simulate 900,000 such objects.
- Runs could be made faster by using event-driven simulation. A 10 fold speed up was achieved this way.
  - Important for long run times over the life of the system.

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