ARROW TECH > If you can't get a bigger target...

6 DOF with Trades Example

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PRODAS 6 Dof with trades example



The following presentation is an example of using the 6 degree of freedom trajectory module within PRODAS with the trade study options. For this example we will run the following:

- 40mm M781 Grenade fired from a M203 Launcher
 - Study the horizontal impact patterns of projectiles fired at 3 different quadrant elevations
 - Introduce an error budget for the following initial conditions Muzzle Velocity, Temperature, Pressure, Winds and Ammunition Dispersion
 - Verify Dispersion Performance buy firing against a 100m vertical target
 - Simulate 3000 separate 6 DOF of freedom trajectories at each Quadrant Elevation

Projectile Model

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40mm M781

Projectile Data



jectile Data					
Projectile Dia.	4.054	cm			
Weight	171.100	gm			
Axial Inertia	393.590027	gm-cm^2			
Trans Inertia	445.42001	gm-cm^2			
CG (from nose)	5.02	cm			
Initial Conditions					
Muzzle Velocity	75.0	m/sec	Initial Pitch Angle +	up 0.000	deg
Initial Spin	385.8	rad/sec	Initial Yaw Angle +	rt 0.0	deg
Quadrant Elevation	28.125	deg	Initial Pitch Rate +	up 30.0	rad/sec
Gun Azimuth (from N)	0.000	deg	Initial Yaw Rate +	rt 30.0	rad/sec
Initial Range	0.0	m			
Initial Deflection	0.0	m			
Initial Altitude (re sl	0.0	m			
Met Data at the Launch	Location				
Gun Altitude above SL	0.0	m	Density	1.2250	kg/m^3
Temperature	15.0	С	Range Wind	0.0	m/sec
Pressure	1013.2	millibars	Cross Wind	0.0	m/sec
Gravity Acceleration	9.80665	m/sec**2			

6DOF With Trades Interface

- PRODAS
File Edit Analysis Support Tools Windows Help
PRODAS I Subject Sub



- Error Budget for the analysis example
 - Muzzle Velocity Error 0.8 m/s (1 std deviation)
 - Ammunition Dispersion 0.5 mils Radial (1 std deviation)
 - An Example or test of dispersion against a vertical target will be run first to determine angular motion needed at the muzzle to generate corresponding aerodynamic jump for the example projectile.
 - Met Errors (all one standard Deviation)
 - 2 deg Celsius
 - 10 millibars
 - 2.75 m/s Cross and Head/Tail Winds
- Will be fired at 3 different quadrant elevations and 3000 trajectories at each range with the error budget super imposed.
- These errors represent error levels for the analysis example
 - User encouraged to use error levels seen in live fire testing along with specific MET errors for specific mission requirements. This is just an example and the error budget ,while trying to remain realistic is just an example and does not represent an official error budget for this system

Example Details – Determining Angular Rate

- The final assessment of the example analysis will be the pattern of the projectile on a horizontal surface. However prior to that we must determine the angular rates needed to produce the desired radial dispersion of 0.5 mils which can only be determined against a Vertical Target down range. To do this we will show examples of the following
 - Simulate firing against a vertical target
 - Stop projectile at desired range
 - Input Error Levels for angular rate at muzzle
 - Run Simulation
 - Assess results
 - Modify input error levels
- Radial Dispersion is defined as

$$\sigma_r = \frac{\left(\sigma_h - \sigma_v\right)}{2}$$

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- For this part of the example we will be assessing projectile performance against a vertical target at 100m distance from the Weapon.
 - Go to the Ballistic Match Module (Analysis Pull down menu, "Firing Tables Ballistic Tables"
 - The following Interface should appear

	ormatted Output Plots		
nalysis Options ———		Projectile Parameters	4/6 DOF Aerodynamic Coefficient Form Factors
Trajectory Simulation Option Use Thrust/Fumer		Flight Weight 171.10000 gm	Axial Force FF 0.920 Normal Force FF 1.000
		Range / Output Control	
siectory / Atmosphere Ir	utialization	Range Output Increment 25.0 m	Roll Damping FF 1.000
Muzzle Velocity	75.0 m/sec	Table Max Range 3000. m	Drag FF During Burn 1.000
Gun Altitude	0.0 m		Fin Cant 1.00
Met Table Source	Std		Thrust Delta Drag Form Factor
Gun Azimuth	0.000 millrad 📐		Delta Drag FF 1.00
Pitching Moment FF	1.00	Induced Pitch Force FF 0.00	
Damping Moment FF	1.00	Induced Side Force FF 0.00	<u>B</u> un Analysis
Magnus Moment FF	1.00	Induced Pitch Mom. FF 0.00	
Side Moment FF	0.00	Induced Side Mom. FF 0.00	
		Induced Roll Mom. FF 0.00	



 Make sure the 6 DOF trajectory option is chosen (vs 4DOF option) and run the simulation. Once complete please open up the Tabular Results interface and view the information at 100m

	5 <u>6 x</u>									
	Range	RemainingV	TimeOfFlight	QE	AngleOfDes	KineticEnerc	Drop	Drift	VertexHeigh	VertexRang
Jnits	m	m/sec	sec	Gunner Mil	deg	kilojoule	deg	deg	m	m
Red=Lock										
	0.0	75.00	0.0000	0.00	0.00	0.5	0.00	0.00	0.0	0.0
	25.0	73.77	0.3362	22.03	1.29	0.5	1.24	0.05	0.1	12.7
	50.0	72.56	0.6787	45.02	2.61	0.5	2.53	0.11	0.6	25.0
	75.0	71.38	1.0278	68.61	4.00	0.4	3.86	0.16	1.3	37.9
	100.0	70.22	1.3842	92.87	5.46	0.4	5.22	0.22	2.3	50.6
	125.0	69.09	1.7493	117.90	7.02	0.4	6.63	0.29	3.7	63.3
	150.0	67.99	2.1244	143.84	8.67	0.4	8.09	0.35	5.5	76.4
	175.0	66.92	2.5102	170.77	10.40	0.4	9.61	0.42	7.7	89.1
	200.0	65.88	2.9082	198.86	12.21	0.4	11.19	0.50	10.4	102.4
	225.0	64.89	3.3202	228.27	14.15	0.4	12.84	0.57	13.5	115.2
	250.0	63.94	3.7497	259.33	16.23	0.3	14.59	0.66	17.2	128.5
	275.0	63.03	4.1996	292.34	18.48	0.3	16.44	0.75	21.6	141.7
	300.0	62.19	4.6736	327.67	20.89	0.3	18.43	0.85	26.8	154.9
	325.0	61.40	5.1800	366.13	23.51	0.3	20.59	0.97	32.9	168.5
	350.0	60.68	5.7269	408.56	26.42	0.3	22.98	1.10	40.2	181.8
	375.0	60.05	6.3323	456.75	29.72	0.3	25.69	1.26	49.1	195.4
	400.0	59.54	7.0305	514.11	33.59	0.3	28.92	1.46	60.6	209.0
	425.0	59.20	7.9080	589.42	38.54	0.3	33.15	1.76	76.6	222.8
	425.0	60.58	11.3254	937.27	58.96	0.3	52.72	3.87	156.8	224.7
	400.0	61.13	11.9160	1012.61	62.94	0.3	56.96	4.63	173.3	212.1
	375.0	61.52	12.3296	1070.23	65.98	0.3	60.20	5.33	185.3	199.6
	350.0	61.79	12.6449	1117.58	68.54	0.3	62.86	6.01	194.6	187.5
	325.0	61.93	12.9000	1157.70	70.88	0.3	65.12	6.63	202.0	176.0
	300.0	61.79	13.1345	1193.24	73.31	0.3	67.12	7.12	208.1	165.0
	275.0	61.30	13.3613	1222.42	75.75	0.3	68.76	7.28	212.9	155.3
	250.0	59.88	13.6705	1250.33	78.41	0.3	70.33	6.91	217.1	145.5
	225.0	56.56	14.1375	1278.76	80.68	0.3	71.93	7.11	221.2	135.1

- A QE of 92.87 mils is needed to impact at 100mm range. We will added 3 mils to this to bias the impact in altitude by 0.3 m along with raising the weapon to 2 meters altitude to ensure that all of the rounds will impact on target. Since in this part of the example we will modifying only the Angular Rates to verify the impact dispersion we now need to set up the trade analysis to stop at 100 m range
- Please open the Analysis pull down menu and go to the Trajectory submenu and choose "Fixed Plane- 6D w/ Trade Studies"
- You should see the Setup Interface appear. Notice slant range is set 100m

/Run Output Setup	Initial Conditions Projecti	le Parameters Aero Form Fac	ctors Formatted Results Tab
mulation Control ——			
Use Thrust/Fumer	🗖 No	Coriolis Option	□ No
Launched from Aircraf	t 🗖 No		
mulation Termination O	ptions	Analysis Control	
Stop at Time	0.0 sec	Integration Interval	20.
Stop at Slant Range	100.		

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- Next go to the Initial Conditions Interface and set the QE to 95.67 Gunner's Mils.
- Set the initial Z position to 2 m altitude (This ensures that all of the impact will be captured at 100m and not fall short)

up/Run Output Setup Initial Conditions Projectile Pa	rameters Aero Form Factors	Formatted R	esults Tabul
	Met		
Input ICs with Table (Use Either u,v,w or Vx,Vy,Vz)	Met Table Source	Std	
	User Met Type	Altitude-B	ased
un Setup	Projectile Initial Position -		
Quadrant Elevation 95,570 Gunner	Initial X position	0.	m
Gun Azimuth 0.000 millrad	Initial Y position	0.	m
	Initial Z position	2.	m 🔊
oiectile Spin Rate	Start at Time	0.000	sec
Muzzle Velocity 75.0 m/sec	Projectile Angles and Bat	a	
Spin at Muzzle 386. rad/sec	Initial Pitch Angle	0.0	deg
Twist 30.49 cal/rev	Initial Yaw Angle	0.0	deg
Exit Spin Ratio	Initial Pitch Rate	0.0	rad/sec
	Initial Yaw Rate	0.0	rad/sec s



- Please save the file at this time via the File pull down menu at the top left of the PRODAS interface.
- Now please go to the trade Interface with the 6D w/Trade Studies Interface

Setup/Run Output Setup Initial Conditions	Projectile Parameters Ae	ero Form Facto	ors Formatte	d Results T	abular Results Scenario Gi	Plotted Re enerator —	sults Trade!	Studies Tra	de Study Error Trade	Budget Tra Studies	de Study Res	cults	
Initial Conditions Trojectile ICs Mass Properties Aerodupamic Form Factors	Spin Options	RODAS aults	Reset to U Nominal V	se All alues	Append to	Existing Tab	le 🗖 No e Scenerios			Exec	ute Trade Stu	idy	
Rocket Motor Atmosphere	C Use Spin Trade Study Si Check for Bir (1)	Values Entere cenario Table Summary of I	nputs										
Rocket Motor Atmosphere	C Use Spin Trade Study S Check for ■ ■ ■	Values Entere cenario Table Summary of I	nputs	IY		QE	Gun Az	Muz Vel	Spin	Pitch Ang	Yaw Ang	Pitch Rate	Ya
Rocket Motor Atmosphere	C Use Spin Trade Study S Check for Bar Carlos for Units	Values Entere cenario Table Summary of I	nputs	Y	Z	QE Gunner Mil	Gun Az millrad	Muz Vel m/sec	Spin rad/sec	Pitch Ang deg	Yaw Ang	Pitch Rate rad/sec	Yav
Rocket Motor Atmosphere	Trade Study So Trade Study So Check for Units Red=Lock	Values Entere cenario Table Summary of I	nputs	Y	Z m User	QE Gunner Mil User	Gun Az millrad	Muz Vel m/sec	Spin rad/sec	Pitch Ang deg	Yaw Ang deg	Pitch Rate rad/sec	Yav

Please use the "Load PRODAS Default" command which will load the saved data from the prior input screens.

 Using the "Trade Space Variables Editor" section please expand the input tree and verify the following values for Z position, Quadrant Elevation and



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- This data was populated by the interface itself when the "Load PRODAS Default" Command was used. The user can overide these if they choose.
- Please use the "Generate Scenarios command to populate the "Trade Study Scenario Table"
- You will now see the Trade Study Scenario Table has been populated with the Initial Conditions Saved in the file.
- This table will be used as our baseline trajectory setup to which we will add errors to.

- 6D w/ I rade	e Studies						12					2:3
rameters Ae	ero Form Facto	ors Formatter	d Results Ta	abular Results	Plotted Re	sults Trade 9	Studies Trac	le Study Error	Budget Tra	ide Study Res	sults	
Global Options Load P Defi Spin Options	RODAS aults Spin from Mu:	Reset to U Nominal Va zzle Velocity a	se All alues	Scenario Go Append to	enerator — Existing Tab Generat	le 🥅 No e Scenerios		Trade	Studies — Exec	ute Trade Stu	ıdy	
ade Study Sc	cenario Table Summary of In ■1 _ — Inor	nputs										
			0		05	Cur An	March Val	C	Direk Ann	Van As	Dital Data	10
Units	Scenario	m	m	∠ m	Gunner Mil	millrad	muz vei m/sec	rad/sec	deq		rad/sec	rad/
RedeLock	1	0.000	0.000	User 2.000	User 95.5700	0.0000	75.00	385.850	0.0000	0.0000	0.0000	j
			PR	ODAS SCEN	ARIO GENEF 1 SCENARIO	NAT X						



- Now the user should go to the "Trade Study-Error Budget Interface"
 - Interface allows input of errors to the input sources available in the Trade Space Variables Error Editor
- In this case we are only interested in the angular rates at launch.
- Please expand the Initial Conditions- Projectile IC Error Tree and choose "Pitch Rate"
- Please Check the "Include this Error in the Trade Study" box and input a 4 rad/sec rate (1 std deviation) in the Pitch Rate Error level box
- Do the same for Yaw Rate
- (See next page for details)







- ARROW TECH > If you can't get a bigger target...
- Please go to the Error Generator section and modify the Number of Errors input to 4000 (This just gives us a lot of errors to minimize random number generator noise)
- Hit the Generate Errors command and you will see the Error Table Populate itself.
- Notice the only errors input to the table are the initial Pitch and Yaw rates at launch.
- If you check the "Check for Input summary and statistics" you will notice the output in the windows summarizes the errors generated for this run

Error Gener	ator Number of f No. Round	Errors to Gene Is/Weapon Genera	erate 4000 te Errors)	Execute	Trade Study (Exect Trade S With Et	with Errors — ute itudy rrors					
Error Table -			-	Check for	Summary of I	nputs and Sta	atistics						
B	3 8 8												
	Error #	IX	Y	z	QE	Gun Az	Muz Vel	Spin	Pitch Ana	Yaw Ang	Pitch Rate	Yaw Rate	Mass II
Units		m	m	m	Gunner Mil	deg	m/sec	rad/sec	deg	deg	rad/sec	rad/sec	gm (
Red=Lock													
	1	0.000	0.000	0.000	0.0000	0.0000	0.00	0.000	0.0000	0.0000	-2.8803	-3.2859	0.00000
	2	0.000	0.000	0.000	0.0000	0.0000	0.00	0.000	0.0000	0.0000	-0.2404	-0.4656	0.00000
	3	0.000	0.000	0.000	0.0000	PRODAS	SCENARIO G	ENERATOR	X	0.0000	-7.1804	7.8779	0.00000
1	4	0.000	0.000	0.000	0.0000	- TRODATO	Jeen and a	Enteronion		0.0000	4.8160	3.3063	0.00000
	5	0.000	0.000	0.000	0.0000				2	0.0000	8.7873	1.4188	0.00000
1	6	0.000	0.000	0.000	0.0000	GENER	TED 4000 SI	TS OF FRRC	IRS	0.0000	-9.7330	0.4243	0.00000
	7	0.000	0.000	0.000	0.0000	GENER	1120 1000 01	LIS OF LINE	1	0.0000	-12.0740	-3.1618	0.00000
	8	0.000	0.000	0.000	0.0000				2	0.0000	0.9960	8.6349	0.00000
	9	0.000	0.000	0.000	0.0000					0.0000	2.2961	4.2092	0.00000
	10	0.000	0.000	0.000	0.0000			C	К	0.000 <mark>0</mark>	1.6789	7,5399	0.00000
	11	0.000	0.000	0.000	0.0000			-		0.0000	2.1921	1.7591	0.00000
	12	0.000	0.000	0.000	0.0000	-0.0000	0.00	0:000	0.0000	0.0000	-2.5396	6.6724	0.00000
	13	0.000	0.000	0.000	0.0000	0.0000	0.00	0.000	0.0000	0.0000	-2.3567	0.9062	0.00000
	14	0.000	0.000	0.000	0.0000	0.0000	0.00	0.000	0.0000	0.0000	-1.8869	-5.4763	0.00000
	15	0.000	0.000	0.000	0.0000	0.0000	0.00	0.000	0.0000	0.0000	0.8737	1.3996	0.00000
	16	0.000	0.000	0.000	0.0000	0.0000	0.00	0.000	0.0000	0.0000	-7.5540	2.4535	0.00000
	17	0.000	0.000	0.000	0.0000	0.0000	0.00	0.000	0.0000	0.0000	0.4559	-7.1167	0.00000
	18	0.000	0.000	0.000	0.0000	0.0000	0.00	0.000	0.0000	0.0000	0.4348	3.0849	0.00000
	1 19	0.000	0.000	0.000	0.000	0.000	0.00	0.000	0.000	0 0000	1 2292	.0 7/19	0.0000







- Now that we have generated our baseline Trade Study and the Trade Study Errors please use the "Execute Trade Study With Errors" command within the "Trade Study Error Budget Interface" window
- Wait for the analysis to finish you may get some VB script warnings about how long the analysis is taking –ignore these.

Spin Pitch Angle Yaw Angle Pitch Rate Yaw Rate Mass Properties Acrodynamic Form Factors Rocket Motor Atmosphere Check for Summary of Inputs and Statistics Check for Summary of Inputs and S	
Rocket Motor Atmosphere Atmosphere Atmosphere Atmosphere Atmosphere Atmosphere Atmosphere Atmosphere Atmosphere Atmosphere Atmosphere Attronometer A	
The following variables were used to generate the initial conditions Error	
	or Table:
Pitch Rate	
hide this Firm in the Trade Shudu? Input Standard Deviation 4.0000 rad/sec	
Std Dev of Generated Errors 4.0329 rad/sec	
n Hate Ves Mean of Generated Errors -0.0015 rad/sec	
Yaw Rate	
Input Standard Deviation 4.0000 rad/sec	
Ind to Round Error in Std Dev of Generated Errors 4.0285 rad/sec	
Wean of Generated Errors 0.0246 rad/sec	



- Once the anlysis is complete we can take a look at the output in the "Trade Study Results interface. The first set of results are in tabular form and show the impact points relative to the nominal trajectory
 - Note the 1st Scenario Number 0 represent the nominal flight of the prjectile without any error

etup/Run	Output Setup	Initial Condit	ions Projecti	ile Parameters	Aero Form F	actors Forr	matted Results	: Tabular Ri	esults Plotte	d Results T	rade Studies	Trade Study	Error Budget	Trade Study Results
Select Result	s Output Type	(Tabulated, F	Plotted, Format	tted): Tabular	Output									
•	3 6 x													
1	Scenario	Error #	X Final	Y Final	Z Final	Velocity	TOF	AOF	Spin	X Miss	Y Miss	Z Miss	R Miss	65
Units	2		m	m	m	m/sec	sec	deg	rad/sec	m	m	m	m	
Red=Lock									1					
	1	(100.00	-0.39	2.26	70.19	1.3846	5.1712	370.2340	0.000	0.000	0.000	0.000	
	1	1	100.00	-0.43	2.30	70.13	1.3853	5.6161	370.2340	0.000	-0.036	0.036	0.051	
	1	2	2 100.00	-0.40	2.27	70.18	1.3847	5.2391	370.2340	0.000	-0.005	0.003	0.006	
	1		3 100.00	-0.28	2.32	69.97	1.3873	4.6898	370.2340	0.000	0.110	0.055	0.123	
	1	1	100.00	-0.36	2.20	70.17	1.3849	4.7685	370.2340	0.000	0.034	-0.066	0.074	
	1 1		100.00	0.00	2.15	70.10	1 2055	E 040E	270 2240	0.000	0.005	0.110	0.110	



• The next set of outputs are the plotted outputs. N this case we are interested in the Vertical Target Impacts so please select that plot.





• The final option for output is the "Formatted Output" which gives our input summary and impact statistics so the we verify the dispersion performance at the angular rate errors input.

			<u></u>		
Range Wind (NEWPAGE)	m/sec	0.000			
4000 Monte Ca	rlo error runs wit	th the following A	error charact	eristics were	e completed
Pitch Rate	rad/sec	-0.00153 4	.03343		
Yaw Rate	rad/sec	0.02455 4	.02898		
(NEWPAGE }		(
NEWPAGE }					
NEWPAGE}	ULTS FOR THE FIRST	1 OF THE	1 SCENARIOS		
NEWPAGE } RADE STUDY RESU Cenario 1	ULTS FOR THE FIRST	1 OF THE	1 SCENARIOS	~	
NEWPAGE} RADE STUDY RESI Cenario 1 Nominal Traje	ULTS FOR THE FIRST	1 OF THE	1 SCENARIOS	and Std Dev	of:
NEWPAGE} RADE STUDY RESI cenario 1 Nominal Traje X 9:	ULTS FOR THE FIRST ctory Terminated a 9.999 m	I OF THE THE Error Runs	1 SCENARIOS had a Mean 100.000	and Std Dev o	of:
NEWPAGE} RADE STUDY RESI cenario 1 Nominal Traje X 9: Y -	ULTS FOR THE FIRST ctory Terminated a 9.999 m 0.392 m	T 1 OF THE	1 SCENARIOS had a Mean 100.000 -0.391	and Std Dev 6 0.001 m 0.050 m	of:
NEWPAGE} RADE STUDY RESI cenario 1 Nominal Traje X 9: Y -(Z - Ting	ULTS FOR THE FIRST ctory Terminated a 9.999 m 0.392 m 2.263 m	I OF THE THE Error Runs X Y Z	1 SCENARIOS had a Mean 100.000 -0.391 2.258 1 205	and Std Dev 0 0.001 m 0.050 m 0.048 m	of:



Trade Study Scenarios Tab

- Text Window shows results of generating an error table or generating scenarios.
- List box to select one of three tables to display
- Scenarios Table
 - Table of IC's to be used in the simulation
 - Each row is a run
- Error Level Table
 - Summary table of all errors and trade space variables that generated the scenarios
- Errors Table
 - Table of error deltas based on the error definitions

		Tar	- ANATI AD		4D undet	- 0							
		11.	ајнатњар 11.	/10/2014	19:20	e - 0							
		S	cenerioG	enerator	Version	1.0							
The worldh	he variable(s) used to generate the trade snace were:												
Ine variab	ue varianie(s) used to generate the trade space were:												
PARAM4	ARAM4 Using the following range:												
	Starting with 0.0000												
	Finishing with 316.0045 With this increment 45.0000												
	• TER ERIC 45.0000												
·				Scenarios									
Scenarios													
Scenarios			~										
Scenarios	3												
Scenarios	Buntt	NominalBun	×	IY.	7	phi	theta	nsi	11	v	w	In	
Scenarios	😼 📙 🔀	NominalRun	× m	Y m	<u>Z</u>	phi deg	theta Gunner Mil	psi deg	u m/sec	v m/sec	w m/sec	p rad/se	
Scenarios	B S	NominalRun Gen	X m Gen	Y m Gen	Z m Gen	phi deg Gen	theta Gunner Mil Gen	psi deg Gen	u m/sec Gen	v m/sec Gen	w m/sec Gen	p rad/se Ge	
Scenarios	B Kun# Gen	NominalRun Gen 12	X m Gen 0.0000	Y m Gen 0.0000	Z m Gen 0.0000	phi deg Gen 0.0000	theta Gunner Mil Gen 799,9998	psi deg Gen 0.0000	u m/sec Gen 200.0000	v m/sec Gen 0.0000	w m/sec Gen 0.0000	p rad/se Ge	
Scenarios	B SS Run# Gen 1 2	NominalRun Gen Ø	X m Gen 0.0000 0.0000	Y m Gen 0.0000 0.0000	Z m Gen 0.0000 0.0000	phi deg Gen 0.0000 0.0000	theta Gunner Mil Gen 799.9998 799.9998	psi deg Gen 0.0000 0.0000	u m/sec Gen 200.0000 200.0000	v m/sec Gen 0.0000 0.0000	w m/sec Gen 0.0000 0.0000	P rad/se Ge	
Scenarios	Bun# Gen 1 2 3	NominalRun Gen Ø Ø	X m 0.0000 0.0000 0.0000	Y Gen 0.0000 0.0000 0.0000	Z Gen 0.0000 0.0000 0.0000	phi deg 0.0000 0.0000 0.0000	theta Gunner Mil Gen 799,9998 799,9998 799,9998	psi deg 0.0000 0.0000 0.0000	u m/sec Gen 200.0000 200.0000 200.0000 200.0000	v Gen 0.0000 0.0000 0.0000	w m/sec Gen 0.0000 0.0000 0.0000	p rad/se Ge 0 0	
Scenarios	B E X Run# Gen 1 2 3 4	NominalRun Gen Ø Ø Ø	X m 0.0000 0.0000 0.0000 0.0000	M Gen 0.0000 0.0000 0.0000 0.0000	Z Gen 0.0000 0.0000 0.0000 0.0000 0.0000	phi deg 0.0000 0.0000 0.0000 0.0000 0.0000	theta Gunner Mil Gen 799.9998 799.9998 799.9998 799.9998 799.9998	psi deg 0.0000 0.0000 0.0000 0.0000 0.0000	u m/sec Gen 200.0000 200.0000 200.0000 200.0000 200.0000	▼ m/sec Gen 0.0000 0.0000 0.0000 0.0000 0.0000	w Gen 0.0000 0.0000 0.0000 0.0000 0.0000	P rad/se G(C C C	
Scenarios	B C C C C C C C C C C C C C C C C C C C	NominalRun Gen Ø Ø Ø Ø	X m Gen 0.0000 0.0000 0.0000 0.0000 0.0000	Y Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	Z Gen 0.0000 0.0000 0.0000 0.0000 0.0000	phi Gen 0.0000 0.0000 0.0000 0.0000 0.0000	theta Gunner Mil Gen 799.9998 799.9998 799.9998 799.9998 799.9998 799.9998	psi Gen 0.0000 0.0000 0.0000 0.0000 0.0000	u m/sec Gen 200.0000 200.0000 200.0000 200.0000 200.0000 200.0000	v Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	w Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	P rad/se Ge C C C C C	
Scenarios	Cen Gen 1 2 3 4 5 6 7	NominalRun Gen Ø Ø Ø Ø Ø	X m Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	Y Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	Z Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	phi Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	theta Gunner Mil Gen 799.9998 799.9998 799.9998 799.9998 799.9998 799.9998	psi deg 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	u m/sec Gen 200.0000 200.0000 200.0000 200.0000 200.0000 200.0000	v Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	w m/sec Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	P rad/se Gr C C C C C C C C C C C C C C C C C C	
Scenarios	Cen Gen 1 2 3 4 5 6 7 8	NominalRun Gen Ø Ø Ø Ø Ø Ø	X m 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	M Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	Z m Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	phi deg 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	theta Gunner Mil Gen 799.9998 799.9998 799.9998 799.9998 799.9998 799.9998 799.9998	psi deg 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	u m/sec Gen 200.0000 200.0000 200.0000 200.0000 200.0000 200.0000 200.0000 200.0000	v m/sec Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	w m/sec 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	P rad/se Ge 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	
Scenarios	Run# Gen 1 2 3 4 5 6 7 8	NominalPun Gen Ø Ø Ø Ø Ø Ø Ø	X m 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	Y Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	Z Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	phi deg Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	theta Gunner Mil Gen 799.9998 799.9998 799.9998 799.9998 799.9998 799.9998 799.9998	psi deg Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	u Gen 200.0000 200.0000 200.0000 200.0000 200.0000 200.0000 200.0000	▼ Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	w Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	P rad/se Gi C C C C C C C C C C C C C C C C C C	
Scenarios	B B XX Run# 1 2 3 4 5 6 7 8 7	NominalRun Gen Ø Ø Ø Ø Ø Ø	X m Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	Y Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	Z Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	phi deg 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	theta Gunner Mil 799, 9938 799, 9938 799, 9938 799, 9938 799, 9938 799, 9938 799, 9938	psi deg 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	u m/sec 200.0000 200.0000 200.0000 200.0000 200.0000 200.0000 200.0000	v Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	w Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	P rad/se G(C C C C C C C C C C C C	
Scenarios	B XX Run#	NominalRun Gen Ø Ø Ø Ø Ø Ø	X m Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	Y Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	Z Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	phi deg Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	theta Gunner Mil Gen 799 9938 799 9938 799 9938 799 9938 799 9938 799 9938 799 9938 799 9938	psi deg Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	u m/sec Gen 200.0000 200.0000 200.0000 200.0000 200.0000 200.0000 200.0000	v m/sec Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	w m/sec Gen 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000	P rad/se Ge C C C C C C C	

Trade Study Results



- Summary data brought back to PRODAS when a trade studies run is complete
- One row for each run

🌆 TrajMATLAB_o	course_4D_update - MATLAB	Trajectory (6DOF-7DOF)			_ 🗆 🗙					
Setup Initial Condit	tions Aero Form Factors 7DOF Sc	uibs Tabular Results Plotted Resu	lts Trade Studies Trade Study Sce	narios Trade Study Results						
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<u>15</u>	TrajMATLAB_course_4D_update - MATLAB Trajectory (6DOF-7DOF)													
S	tun İ Initial C	onditions A	ero Form Fa	ctors 7D0F	Squibs Tab	ular Besults	Plotted Resu	lts Trade St	udies Trade	Study Scena	rios Trade 9	Study Results		
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1	Show Plot	s												
i			×	Y	7	velocitu	TOF	AOF	۵۵۵	Spin	DBmiss	CBmiss	BadMiss	Para
	Units	10011	m	m	m	m/sec	sec	deg	deq	rad/sec	m	m	m	- Gire
	Red=Lock													
		1	4083.	-2.4	0.00	169.54	31.0411	58.8009	4.0730	69.1535	0.000	0.000	0.000	0.
		2	3861.	77 297.2	7 0.00	170.92	29.8678	52.8925	4.1264	69.4532	0.000	0.000	0.000	0.
		3	3531.	70 297.7	3 0.00	173.01	27.9614	40.1733	4.1321	69.8141	0.000	0.000	0.000	0.
		4	3343.	50 160.7	0.00	174.26	26.7646	29.2802	4.1849	69.9739	0.000	0.000	0.000	0.
		5	3287.	37 -1.8	6 0.00	174.57	26.3980	25.4078	4.1766	69.9942	0.000	0.000	0.000	0.
		6	3343.	16 -164.7	3 0.00	1/4.10	26.7819	29.3338	4.2002	69.9148	0.000	0.000	0.000	U.
		/	3531.	N -302.4	4 0.00	1/2.//	27.9943	40.2718	4.1214	69.7160	0.000	0.000	0.000	U.
	x		3061.	94 ·302.2	5 0.00	170.71	23.3004	53.0663	4.2408	63.3661	0.000	0.000	0.000	0.
	•													-

 From the Results we can see that our standard deviation in Y is .050 m or 0.5 mils and our standard deviation in Z is 0.048m or .48 mils which is ~ 0.49 mils Radial Dispersion. Please increase the initial std deviation in Pitch Rate to 4.3 rad/sec in the "Trade Study-Error Budget Interface". This will result in a std deviation I impact in Z of .050m which equates to 0.5 mils dispersion and the desired radial ammunition dispersion of 0.5 mils

$$\sigma_r = \frac{\left(\sigma_h - \sigma_v\right)}{2}$$

lf vou can't aet a biaaer taraet...



the second care and the second care and	4000	Monte	Carlo	error	runs	with	the	following	error	characteristics	were	completed	for	each	scenari
---	------	-------	-------	-------	------	------	-----	-----------	-------	-----------------	------	-----------	-----	------	---------

		Mean Shift	Std Dev
Pitch Rate	rad/sec	0.05779	4.21041
Yaw Rate	rad/sec	-0.06350	4.03027
{NEWPAGE }			

TRADE ST	UDY RESULTS	FOR THE FIRST	1 OF THE	1 SCENARIOS	
Scenario	1				
Nomina	l Trajectory	Terminated at:	Error Runs	had a Mean and	Std Dev of:
v	00 000	-	v	100 000	0 001 -

X	99.999	m	X	100.000	0.001	m
Y	-0.392	m	Y	-0.392	0.050	m
Z	2.263	m	Z	2.257	0.050	m
Time	1.385	sec	Time	1.385	0.001	sec



Initializing full error budget

- ARROW TECH > If you can't get a bigger target...
- Now that we have verified the ammunition dispersion against a vertical target performance we can now add the other error contributions and now evaluate against a horizontal or ground impact scenario.
- Error Budget for the analysis example
 - Muzzle Velocity Error 0.8 m/s (1 std deviation)
 - Ammunition Dispersion 0.5 mils Radial (1 std deviation)
 - An Example or test of dispersion against a vertical target will be run first to determine angular motion needed at the muzzle to generate corresponding aerodynamic jump for the example projectile.
 - Met Errors (all one standard Deviation)
 - 2 deg Celsius
 - 10 millibars
 - 2.75 m/s Cross and Head/Tail Winds
 - Will be fired at 3 different quadrant elevations and 1000 trajectories at each range with the error budget super imposed.



• Since we are now interested in evaluating the ground impact patterns at 3 different Quadrant Elevations we need to modify the simulation termination options within the Setup interface window. Please modify the slant range input from 100 to 0. This ensures that the rounds will fly to ground impact regardless of the input QE.

Use Thrust/Fumer	□ No	Coriolis Option	□ No
Launched from Aircraft	∏ No		
mulation Termination Opti	ions	Analysis Control	
Stop at Time	0.0 sec	Integration Interval	20.
Stop at Slant Range	0. m		



- The next step is to add the 3 QE's desired to the Trade Study Scenario Tables found under the Trade Study Interface tab. To do this please open the "Trade Study" interface tab and go to the "Trade Space Variable Editor" and open the Weapon Position input tree and highlight Quadrant Elevation.
- There is a pull down menu midway in the interface which allow you to choose the input method. These include "Nominal" (value saved with the file) "Range and Increment" (Allows user to set the initial QE and then add incremantal adjustments) or "Table of Values". Please choose this last option and input the values shown to the right.
- Please use the Generate Scenarios Command to repopulate the Trade Study Scenario Tables



rade space valiable Editor	1	Global Options	-			Scenario G	ienerator —				Studies ——			1	
Initial Conditions		Load PF Defa Spin Options -	ODAS ults	Reset to U Nominal V	Ise All alues and Twist	Append t	o Existing Tab General	le 🗌 No le Scenerios			Exec	ute Trade Sti	ıdy		
Gudadit Elevation Gun Azimuth	90 24	Trade Study Sc	enario Table Summary of I	nputs											
Rocket Motor		B C <i>E</i>	8 8												
Aunosphere			Scenario	X	Y	Z	QE	Gun Az	Muz Vel	Spin	Pitch Ang	Yaw Ang	Pitch Rate	Yaw Rate	Mass
		Red=Lock	User	m	m	m User	Gunner Mil- User	millrad	m/sec	rad/sec	deg	deg	rad/sec	rad/sec	gm
			1	0.000) 0.00() 0.00() 2.00) 2.00) 99.9999) 350.0003	0.0000) 75.00) 75.00	385.850 385.850	0.0000 0.0000	0.0000	0.0000	0.0000	171.10 171.10
Use Table of Values	*	×	3	0.000	0.000	2.00	624.9997	0.0000	75.00	385.850	0.0000	0.0000	0.0000	0.0000	171.10
Quadrant Elevation 100.0000 0 Quadrant Elevation 350.0000 0 Quadrant Elevation 625.0000 0 Quadrant Elevation 625.0000 0	iunner	aria Tabla						PF	RODAS SCEN	ARIO GENEF 3 SCENARIC	RAT X				
Quadrant Elevation 0.0000	Check for Su	immary of Inputs									OK				
Quadrant Elevation 0.0000	he variabl	40 .e(s) used to	nm 781 6 Sceneri generate	Dof With 03/11/201 oGenerato the trad	Trades Exa 5 09:29 r Version e space we	amp - 0 1.0 ere:									

ARROW TECH > If you can't get a bigger target...

- From the newly generated "Trade Study Scenario Tables" we can now see the added Quandrant Elevations along with the initial Z of the weapon and Muzzle Velocity we would like to use.
- At this point we should now go to the Trade Study Error Budget interface and go to the Trade Space Variable Error Editor and expand the Initial Conditions, Projectile IC's tree and select the Muzzle Velocity Input and add the 0.8 m/s error. Also modify the number of errors to 3000 instead of 4000 (Total number of trajectories is 10000 per study. When these are combined with the 3 Scenarios this will generate 9000 trajectories.





• To utilize and input MET errors the user has to do 3 things. The first is to input the actual MET errors using the Trade Study Error Budget interface and Trade Space Variable Error Editor and expand the Atmosphere Error Tree. When the user selects Temp a pop up window will appear explaining the next two steps the user will need to do to get these errors in to the Trade Study. (see next Page)

Trade Space Variable Error Editor —————	Error Genera	ator ——						- Execute	Trade Study	with Errors –	_				
Initial Conditions Mass Properties Aerodynamic Form Factors Bocket Mator		Number of No. Round	Errors to ds/Weap	General	e 3000				Exec Trade	ute Study					
Atmosphere Demograture			G	enerate l	Errors					.mors					
Cross Wind	Error Table —	Error Table Check for Summary of Inputs and Statistics													
60 m3		3 <u>8</u> x			.15				10			10			
		Error #	Х	Y	Z		QE	Gun Az	Muz Vel	Spin	Pitch Ang	Yaw Ang	Pitch Rate	Yaw Rate	Mass
	Units	1	m	m	m		Gunner Mil	deg	m/sec	rad/sec	deg	deg	rad/sec	rad/sec	gm
	Red=Lock	12													
			1	0.000	0.000	0.000	J U.UUUU		0.0	J U.UUU			1.2920	-2.1925	
			2	0.000	0.000	0.000	0.0000		0.0	0.000) 3.2878 N 4.9792	0.6261	2 0.00000
		1	4	0.000	0.000	0.000	0.0000	0.0000	0.0	0.000	0.0000	0.0000) -4.3703 E 7960	-2.07.52	. 0.00000
		F	5	0.000	0.000	0.0							X	-2 2307	7 0.00000
		Ê Î	6	0.000	0.000	0.0						A REPORT OF A	<u></u>	-4 4771	0.00000
Include this Error in the Trade Study?		ti s	7	0.000	0.000	0.0								4,7453	3 0.00000
Temp (C)		1 8	3	0.000	0.000	0.0	IN ORDER	TO INCLUDE	ATMOSPH	ERE VARIABI	ES IN THE T	RADE STUDY	: USER	7.8057	2 0.00000
		9	Э	0.000	0.000	0.0	MET MUST	BE SELECTE	D ON THE	NITAL CON	DITIONS TAE	, THE MET 1	TABLE	5.6630	0.00000
		10	0	0.000	0.000	0.0	MUST BE G	ENERATED.	AND MET D	ATA MUST E	E TRANSFER	RED TO THE	TRADE	2.2500	0.00000
Round to Round Error in	Ū.	11	1	0.000	0.000	0.0	STUDY SCE	NARIO TAB	LE BY LOAD	ING PRODA	S DEFAULTS	AND GENER	ATING	-0.9802	2 0.00000
Temp (C) 2 0000	Ū.	12	2	0.000	0.000	0.0	SCENARIO	S.						0.9081	i 0.00000
		13	3	0.000	0.000	0.0								-4.7793	3 0.00000
		14	4	0.000	0.000	0.0								4.9237	/ 0.00000
		15	5	0.000	0.000	0.0							OK	-5.4219	3 0.00000
		16	6	0.000	0.000	0.0						L		6.5696	i 0.00000
		17	7	0.000	0.000	0.0			1					-1.5945	i 0.00000
		18	3	0.000	0.000	0.000	0.0000	0.0000	0.0	0.000	0.0000	0.0000	-1.1763	-5.4986	i 0.00000
		19	3	0.000	0.000	0.000	0.0000	0.0000	0.0	0.000	0.0000	0.0000) 2.8959	5.9203	3 0.00000
	۱ <u> </u>	20]	0.000	0.000	0.000	0.0000		0.0	0.000	0.0000	0.0000	2.8722	-6.1302	2 0.00000
		21		0.000	0.000	0.000	0.0000			0.000			-2.9888	3.6198	
		24	2	0.000	0.000	0.000	0.0000		0.0	0.000			0 4007	•7.7686	0.00000
		2.	5	0.000	0.000	0.000	0.0000		0.0	0.000			0.423/	0.8369	
		24	+	0.000	0.000	0.000	0.0000		0.0	0.000			1.862/	-0.4231	0.00000
		20	2	0.000	0.000	0.000	0.0000		0.0	0.000			1.36/8	3.1613	
		20	7	0.000	0.000	0.000	0.0000) 0.0) 0.0	0.000			-4.4400 7 6060	-0.2362	. 0.00000
		2/	0	0.000	0.000	0.000	0.0000) 0.0) 0.0	0.000			7.0303	-1.7244	2 0.00000
	1	20	81 i	0.000	0.000	0.000	0.0000	0.0000	0.0	0.000	0.0000	0.0000	-2.3471	-3.4332	. 0.00000

- ARROW TECH > If you can't get a bigger target...
- Please Insert the Temp, Pressure and Wind Errors maing sure the Include Error box is checked for each

Trade Space Variable Error Editor Initial Conditions Mass Properties Aerodynamic Form Factors Rocket Motor Atmosphere Cross Wind Range Wind	Setup/Run Output Setup Initial Conditions Projectile Trade Space Variable Error Editor Initial Conditions Mass Properties Aerodynamic Form Factors Rocket Motor Atmosphere Pressure Cross Wind Range Wind	Setup/Run Output Setup Initial Conditions Projectile P Trade Space Variable Error Editor Initial Conditions Mass Properties Aerodynamic Form Factors Rocket Motor Atmosphere Temperature Pressure Cross Wind Range Wind	Trade Space Variable Error Editor
Include this Error in the Trade Study? Temp (C) Round to Round Errer in Temp (C) 2.0000	 Include this Error in the Trade Study? Baro Pressure Round to Round Error in Baro Pressure 10.0000 millibars 	Include this Error in the Trade Study Cross Wind Round to Round Error in Cross Wind 2.750	Include this Error in the Trade Study? Range Wind



- Prior to generating Errors please do the following
 - Go the Initial Conditions interface tab and set the MET source to User
 - Please open the MET Table Genarator (Found under "Support" pull down menu "Met" and generate a std day atmosphere with Sea Leval Temp set to 15C and Pressure set to 1013.25 and hit the generate MET command. This will populate the table.
 - Please Save the PRODAS Model at this point

🏂 40mm 781 6 Dof With Trades Example Fixed Plane	- 6D w/Trade Studies	👫 40mm 781 6 Dof With Trades Example Met
Setup/Run Output Setup Initial Conditions Projectile Pa	rameters Aero Form Factors Formatted Results Tabular Result:	ults Altitude-Based User Met Table Range-Based User Met Table
☐ Input ICs with Table (Use Either u,v,w or Vx,Vy,Vz)	Met	Inputs To Generate an Alikude-Base User Met Table Atmos Temp 15.0 Atmos Pressure 1013.3 milloars Alikude-Temperature Pressure WindVelocit WindAzimutt Inits n C m Milloars Mil
Gun Setup	Projectile Initial Position	Above sea level 0, m Red=Lock 000 15 c 1020 1 2.2 20200
Quadrant Elevation 95.570 Gunner	Initial X position 0. m	Wind Direction 0.0 deg 0.0 100. 100. 100. 3.3 262,00 0.0 deg 0. 14.3 1014.2 3.3 262,00
	Initial Z position 2. m	Wind Velocity 0.00 m/set 100. 13.7 1002.3 3.3 262.00 2000. 13.0 990.4 3.3 262.00 300. 12.4 978.7 3.3 262.00
Projectile Spin Rate	Start at Time 0.000 sec	Comment 400. 11.8 967.1 3.3 262.00 500. 11.2 955.5 3.3 262.00
Muzzle Velocity 75.0 m/sec	Projectile Angles and Rates	Generate Altitude-Based User Met Table 1000. 8.0 899.6 3.3 262.00
Spin at Muzzle 386. rad/sec	Initial Pitch Angle 0.0 deg	1250. 6.4 872.7 3.3 262.00
Twist 30.49 cal/rev	Initial Yaw Angle 0.0 deg	
Exit Spin Ratio	Initial Pitch Rate 0.0 rad/sec s	25001.6 747.6 3.3 262.00
	Initial Yaw Bate	3000, -4.8 /01.8 3.3 262.00 3500 -8.0 658.4 3.3 262.00
Calculate Spin	J c.o have a	400011.3 617.1 3.3 262.00
		6000 -24.2 472.4 3.3 262.09
		7000. 30.7 411.4 3.3 26200 ▼



- Please return to the Trade Study Interface and use the "Load PRODAS Defaults Command.
 - This loads up the saved User MET Table that the errors produced and in the Trade Study Error Budget/Error Table will be applied.



- Regenerate Scenario Table Also (Ensures all of the PRODAS Defaults are loaded)
- Please return to the Trade Study Error Budget interface and use the Generate Errors Command to repopulate the table with the new Muzzle Velocity and Atmospheric Errors Input Previously.



• These error will now appear in the Table and Summary Input for the Table



1	Muz Vel	Spin	Pitch Ang	Yaw Ang	Pitch Rate	Yaw Rate	Mas gm
20	m/sec	rad/sec	deg	deg	rad/sec	rad/sec	
5	-0.06	0.000	0.0000	0.0000	1.3317	4.7474	
Ĵ	-0.95	0.000	0.0000	0.0000	1.4583	-8.5974	
D	-1.00	0.000	0.0000	0.0000	-9.4828	10.2070	
D	0.22	0.000	0.0000	0.0000	6.9210	-3.1288	
D	-0.74	0.000	0.0000	0.0000	-0.7988	1.3505	
D	0.47	0.000	0.0000	0.0000	1.3090	1.8594	
10	2027	2 2 2 2 2			St 202.7		

im	Temp (C)	Pressure	Cross Wind	Range Wind		
		millibars	m/sec	m/sec		
00	2.1844	13.1624	10.040	0.546		
00	-4.0400	-11.7291	2.132	-0.475		
00	4.0815	-0.5093	-2.269	3.755		
00	-2.6737	-6.3032	4.612	-1.330		
00	-2.1496	7.9107	-1.879	0.659		
00	-1.4807	-18.6557	-1.671	1.817		
00	2.6368	19.6137	2.717	-2.526		
00	0.7940	-15.4604	-1.916	-1.794		
00	-3.1636	-3.6488	2.036	0.921		



• Summary of Inputs

40mm 781 6 Dof With Trades Examp - 0

03/11/2015 10:16

ScenerioGenerator Version 1.0

Errors were generated to simulate firing of 3000 rounds.

The following variables were used to generate the initial conditions Error Table:

Muzzle Velocity

Input Standard Deviation	0.8000	m/sec	Temp (deg C)
Std Dev of Generated Error:	s 0.8166	m/sec	Input Standard Deviation 2.0000
Mean of Generated Errors	0.0117	m/sec	Std Dev of Generated Errors1.9690Mean of Generated Errors-0.0378
			Baro Pressure
Pitch Rate			Input Standard Deviation 10.0000 millibars
Input Standard Deviation	4.3000	rad/sec	Std Dev of Generated Errors 9.9646 millibars
Std Dev of Generated Error:	s 4.3045	rad/sec	Mean of Generated Errors 0.0149 millibars
Mean of Generated Errors	-0.0520	rad/sec	Cross Wind
			Input Standard Deviation 2.7500 m/sec
			Std Dev of Generated Errors 2.8143 m/sec
Yaw Rate			Mean of Generated Errors -0.0347 m/sec
Input Standard Deviation	4.0000	rad/sec	Range Wind
Std Dev of Generated Error:	s 3.9172	rad/sec	Input Standard Deviation 2.7500 m/sec
Mean of Generated Errors	0.0654	rad/sec	Std Dev of Generated Errors 2.7088 m/sec Mean of Generated Errors -0.0706 m/sec



- At this point we have done the following:
 - Set up 3 Quadrant Elevations to simulate the input Errors and loaded in the baseline Muzzle Velocity and Atmosphere
 - Verified the Ammunition Dispersion input
 - Generated 3000 initial 1 standard devaition launch errors which will be used for the execution of the analysis which include
 - 1 std deviation in Muzzle Velocity of 0.8 M/s
 - Radial dispersion of 0.5 mils (1 std deviation)
 - Temperature std deviation of 2 deg C
 - Pressure std deviation of 10 millibars
 - 2.75 m/s of Cross and Range winds
- We are now ready to execute the study so please go the Trade Study Error Budget interface and use the Execute Trade Study with Errors command



Results



- Once the analysis has been completed please go to the "Trade Study Results" interface where we can see the tabular output, the plotted output and the summary formatted output
 - Tabular Output summarizes the impacts and miss distance per error and scenario
 - The Plotted Output show the ground impact patterns for each scenario with input error budget (Please choose Ground Impact vs Vertical Impact)
 - Summary of Formatted output summarizes all fo the inputs used ans then shows the average impact and std devaition of the impacts due to the input error budgets

BA A AX

Setup/Run Output Setup Initial Conditions Projectile Parameters Aero Form Factors Formatted Results Tabular Results Plotted Results Trade Studies Trade Study Error Budget Trade Study Results

Select Results Output Type (Tabulated, Plotted, Formatted): Tabular Output

	Scenario	Error #	X Final	Y Final	Z Final	Velocity	TOF	AOF	Spin	× Miss	Y Miss	Z Miss	R Miss
Units	54.	1	m	m	m	m/sec	sec	deg	rad/sec	m	m	m	m
Red=Lock													
(1	0	123.78	-0.61	0.00	69.40	1.7285	7.4946	366.5920	0.000	0.000	0.000	0.000
	1	1	125.52	-0.76	0.00	70.13	1.7366	7.4862	369.5910	1.737	-0.152	0.000	1.743
	1	2	116.97	-0.98	0.00	67.88	1.6635	10.1345	362.7350	-6.816	-0.363	0.000	6.826
(1	3	121.62	-0.38	0.00	68.52	1.7201	7.7209	362.5650	-2.163	0.231	0.000	2.175
	1	4	125.85	-0.43	0.00	68.58	1.7623	6.7771	368.0020	2.068	0.178	0.000	2.076
(1	5	120.41	-0.73	0.00	68.78	1.6977	8.8579	363.1660	-3.372	-0.114	0.000	3.374
(1	6	123.04	-0.56	0.00	68.83	1.7303	7.0996	364.4160	-0.747	0.056	0.000	0.749
(1	7	126.37	-0.38	0.00	69.47	1.7582	6.7384	368.8940	2.590	0.229	0.000	2.600
ĺ.	1	8	124.93	-0.31	0.00	69.07	1.7515	7.0362	366.3150	1.149	0.300	0.000	1.187
1	1	9	121.79	-0.60	0.00	68.78	1.7193	7.6954	362.7430	-1.990	0.010	0.000	1.990
1	1	10	125.61	-0.35	0.00	69.02	1.7623	8.0035	366.4370	1.824	0.265	0.000	1.843
(1	11	124.99	-0.53	0.00	69.47	1.7443	6.9437	367.2050	1.209	0.079	0.000	1.211
Î.	1	12	122.82	-0.27	0.00	68.22	1.7390	7.0560	362.8460	-0.962	0.343	0.000	1.021
ĺ.	1	13	125.05	-0.49	0.00	68.80	1.7517	9.1232	366.9690	1.263	0.121	0.000	1.268
1	1	14	124.76	0.00	0.00	68.11	1.7665	5.5255	364.5820	0.977	0.610	0.000	1.152
(1	15	122.88	-1.26	0.00	69.68	1.6995	9.5688	372.8620	-0.907	-0.648	0.000	1.115
1	1	16	124.63	-0.35	0.00	69.05	1.7481	7.0560	365.7590	0.847	0.260	0.000	0.886
	1	17	117.88	-0.99	0.00	68.29	1.6707	9.2273	362.6640	-5.902	-0.375	0.000	5.914
1	1	18	126.28	-0.73	0.00	70.12	1.7457	7.5515	370.0690	2.495	-0.113	0.000	2.498
(1	19	123.63	-0.49	0.00	68.95	1.7392	8.8562	364.3750	-0.158	0.119	0.000	0.198
	1	20	121.83	-0.61	0.00	68.72	1.7167	7.0225	363.6500	-1.950	0.001	0.000	1.950
(1	21	126.46	-0.17	0.00	68.74	1.7703	9.5785	367.9550	2.678	0.443	0.000	2.714
	1	22	119.61	-0.50	0.00	68.03	1.7035	7.3050	359.6860	-4.179	0.111	0.000	4.180
(1	23	121.72	-0.57	0.00	68.88	1.7169	8.5256	363.0590	-2.065	0.045	0.000	2.066
(1	24	118.57	-0.55	0.00	68.48	1.6831	7.6834	361.0650	-5.219	0.061	0.000	5.219
(1	25	126.98	-0.36	0.00	69.80	1.7612	6.7169	369.6280	3.201	0.256	0.000	3.211
(1	26	124.14	-0.92	0.00	70.05	1.7215	7.6689	369.0800	0.352	-0.303	0.000	0.465
1	1	27	124.08	-0.78	0.00	69.65	1.7205	7.0717	369.7320	0.291	-0.172	0.000	0.338
	1	28	126.53	-0.54	0.00	70.01	1.7520	7.9231	369.7560	2.745	0.069	0.000	2.746
-		1.	1		1								

Plotted Output





Formatted Summary



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{LANDSCAPE}									
•		4 Omm	781 6 Dof	With Trade	s Examp - 0				
			03/	11/2015 10:	48				
		Т	raj20006D_	wTrades Ver	sion 3.1.0				
A trade study has bee	n completed wi	.th 3 s	cenarios.	The follow	ing inputs we	ere used for	the first	3 of the	3 scenarios
Parameter	Units	Scen 1	Scen 2	Scen 3					
Weapon Position									
X	m	0.000	0.000	0.000					
Y	m	0.000	0.000	0.000					
Z	m	2.000	2.000	2.000					
Quadrant Elevation	Gunner Mil	100.000	350.000	624.999					
Azimuth	deg	0.000	0.000	0.000					
Projectile ICs	1.50								
Muzzle Velocity	m/sec	75.000	75.000	75.000					
Spin	rad/sec	385.850	385.850	385.850					
Pitch Angle	deg	0.000	0.000	0.000					
Yaw Angle	deg	0.000	0.000	0.000					
Pitch Rate	rad/sec	0.000	0.000	0.000					
Yaw Rate	rad/sec	0.000	0.000	0.000					
Mass Properties									
Mass	qm	171.100	171.100	171.100					
Azial Inertia	gm-cm^2	393.590	393.590	393.590					
Transverse Inertia	gm-cm^2	445.420	445.420	445.420					
CG Location	cm	5.023	5.023	5.023					
Aerodynamic Form Fact	ors								
Axial Force FF		0.920	0.920	0.920					
Normal Force FF		1.000	1.000	1.000					
Spin Decay FF		1.000	1.000	1.000					
Pitching Moment FF		1.000	1.000	1.000					
Damping Moment FF		1.000	1.000	1.000					
Magnus Moment FF		1.000	1.000	1.000					
Side Moment FF		0.000	0.000	0.000					
Trim Angle (Deg)		0.000	0.000	0.000					
Trim Orientation	deg	0.000	0.000	0.000					
Fin Cant (Deg)		1.000	1.000	1.000					
Rocket Motor									
Thrust FF		1.000	1.000	1.000					
Burn-On Time	sec	0.000	0.000	0.000					
Atmosphere									
Temp (Deg C)		15.000	15.000	15.000					
Baro Pressure	millibars	1013.250	1013.250	1013.250					
Cross Wind	m/sec	0.000	0.000	0.000					
Range Wind	m/sec	0.000	0.000	0.000					
{NEWPAGE }									



 Setup/Run
 Output Setup
 Initial Conditions
 Projectile Parameters
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INTURE		1.12	T.000	1.000	1.00	0				
Burn-On	Time	sec	0.000	0.000	0.00	0				
Atmospher	e									
Temp (D	eg C)	1100000	15.000	15.000	15.00	10				
Baro Pr	essure	millibars	1013.250	1013.250	1013.25	0				
Cross W	ind	m/sec	0.000	0.000	0.00	10				
Range W {NEWPAGE}	ind	m/sec	0.000	0.000	0.00	0				
3000 Mo	nte Carlo e	rror runs with	h the follow	ving error	characte	ristics W	vere com	pleted f	for each	scenario.
			Mean Shif	t Std D	ev					
Muzzle Ve	locity	m/sec	-0.00526	0.825	82					
Pitch Rat	e	rad/sec	-0.06905	4.358	81					
Yaw Rate		rad/sec	-0.11604	4.032	04					
Temp (Deg	C)		-0.04731	1.987	34					
Baro Pres	sure	millibars	0.01172	9.935	53					
Cross Win	d	m/sec	-0.06219	2.737	76					
Range Win {NEWPAGE}	d 📕	m/sec	-0.00204	2.669	93					
Scenario Nominal	1 Trajectory	Terminated at	t: Error	Runs had	a Mean a	nd Std De	v of:			
x	123.784	m	2	12	3.579	2.719	m			
Y	-0.612	m	7		0.596	0.221	m			
Z	0.000	m	2		0.000	0.000	m			
Time	1.728	sec	c	lime	1.729	0.026	sec			
Scenario	2									
Nominal	Trajectory	Terminated at	t: Erron	Runs had	a Mean a	nd Std De	v of:			
х	319.511	m	2	31	8.126	6.865	m			
Y	-5.213	m	2		5.127	1.519	m			
Z	0.000	m	2	2	0.000	0.000	m			
Time	5.053	sec	2	lime	5.045	0.055	sec			
Scenario	3									
Nominal	Trajectory	Terminated at	t: Error	Runs had	a Mean a	nd Std De	v of:			
х	435.782	m	3	43	3.245	10.247	m			
Y	-14.632	m	3	-1	4.435	3.544	m			
Z	0.000	m	2	:	0.000	0.000	m			
Time	8.358	sec	3	ime	8.338	0.086	sec			