

Temperature Insensitive Propulsion for KE Tank Ammunition

Author: Neal Hylton - General Dynamics Ordnance & Tactical Systems:
Large Caliber Ammunition

Co-author: Dr. Klaus-Achim Kratzsch – Rheinmetall Waffe Munition

DISTRIBUTION STATEMENT A. Approved for Public Release; distribution unlimited



LEADING THE WAY

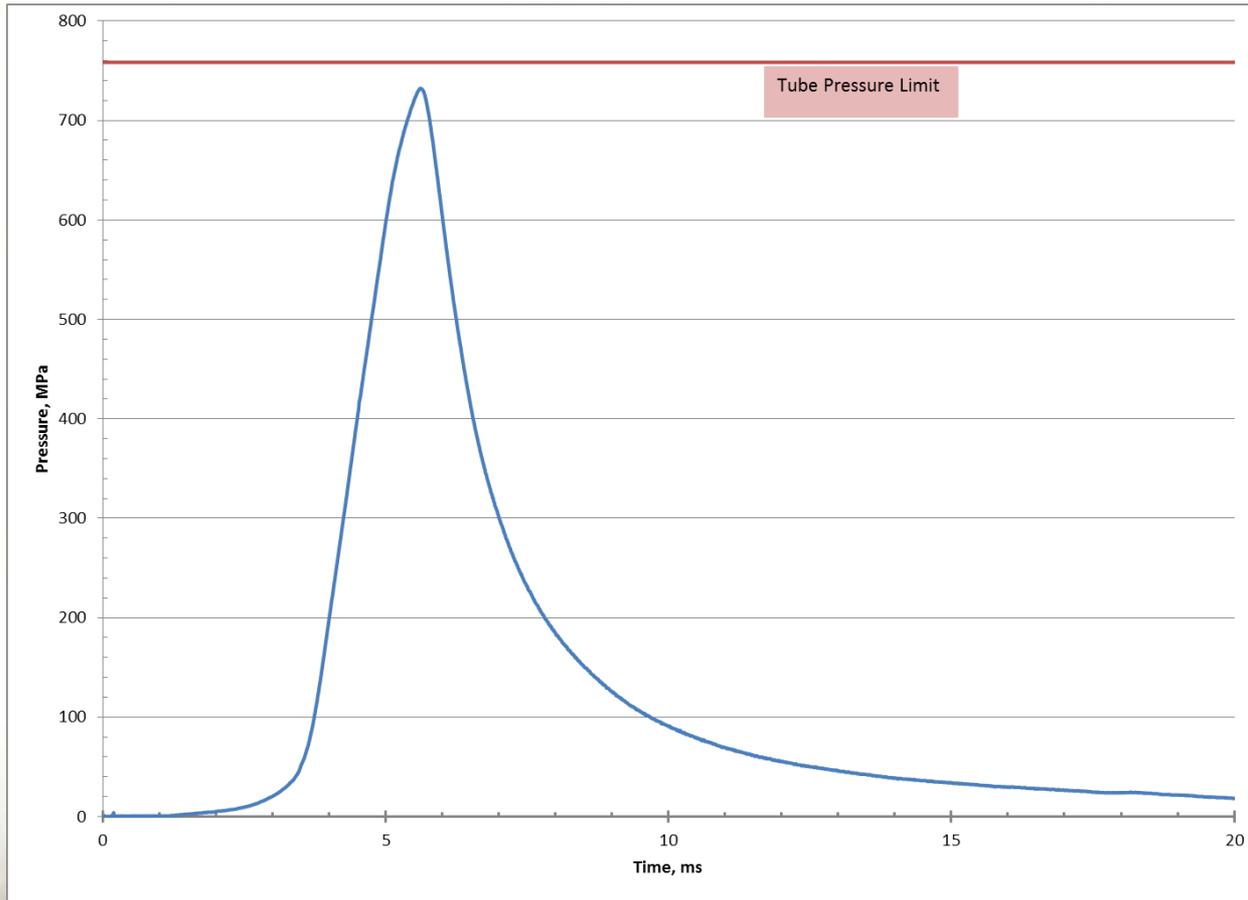
KE Performance Limitations

- $KE = \frac{1}{2}mv^2$
- Improved performance is achieved via increased velocity at target
- Therefore, the main goal is increased muzzle velocity
- Higher velocity is typically obtained through increased operating pressures
- *But* – Cannon pressure limits are established to ensure safety
- Problem – current systems already operate at the maximum safe limit of the cannon (rules out increased velocity via higher operating pressure)



Typical Pressure Curve for KE

- Total energy imparted to the projectile is equivalent to the integrated area under the pressure time curve



LEADING THE WAY

How to Increase Performance

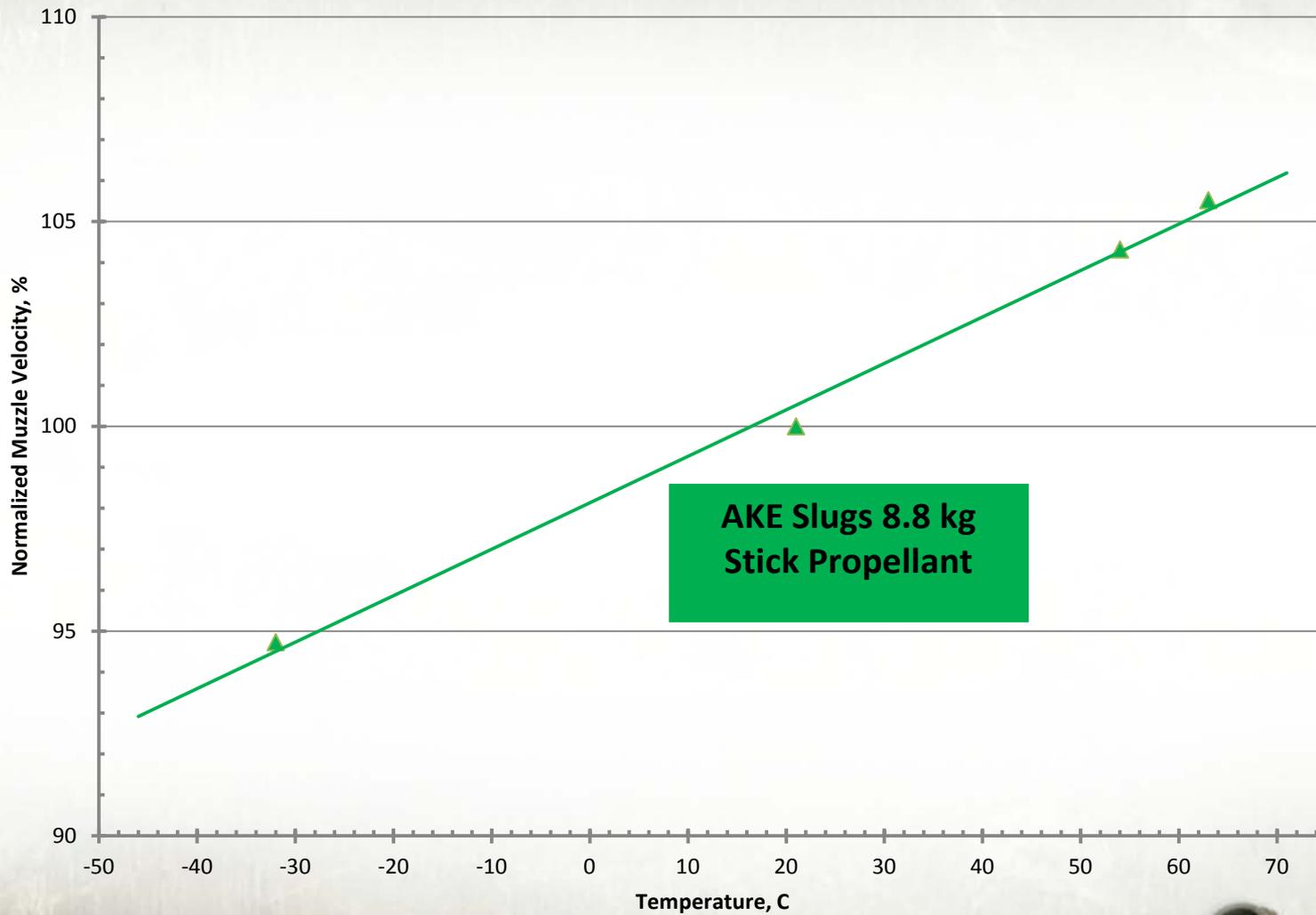
- Many ideas to increase the area without increasing P-max have been limited in successes
 - Programmed Splitting Slotted Stick
 - Multi-layered Propellants
 - Deterred Propellants
- Some successful tests have been performed but not yet fielded due to various issues of safety, producibility, and aging



LEADING THE WAY



Typical Propellant Temperature Profile



LEADING THE WAY

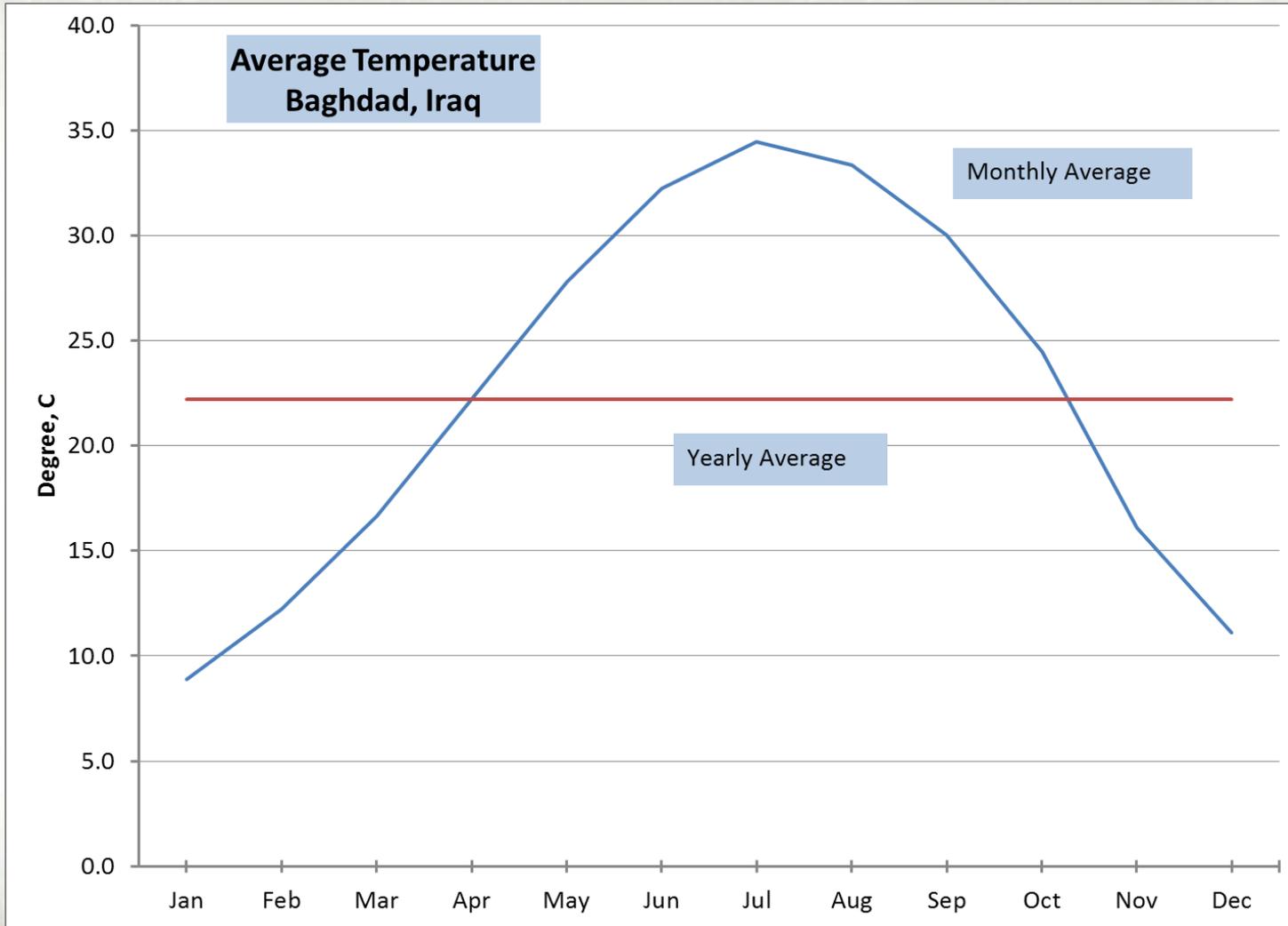
Increase Performance by Altering Temperature Profile

- Every fielded round needs to be capable of firing across the operational temperature range, including extremes
- But – Majority of firing engagements are more normal “ambient” temperatures
- Temperature Insensitive Propulsion System (TIPS®) breaks the normal increased temperature – increased pressure correlation
- This allows for an increase in performance at those temperatures most commonly encountered



LEADING THE WAY

Typical Average Temperature



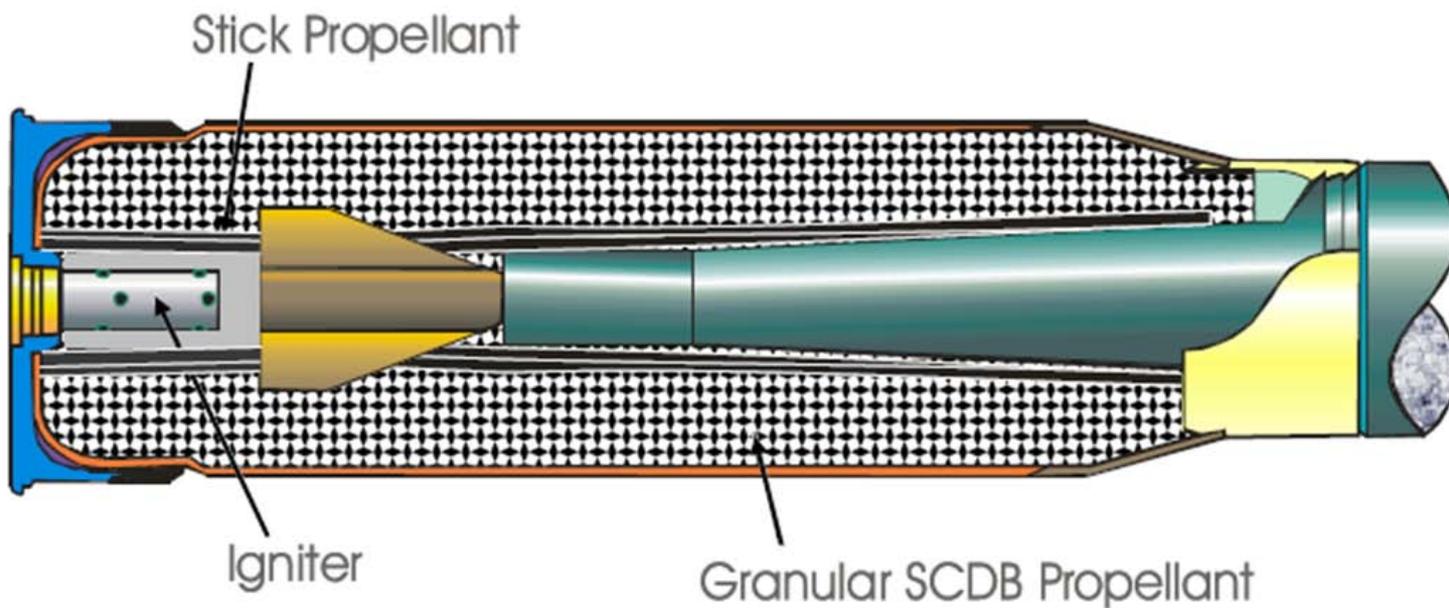
Advanced Propulsion Work

- Ballistic testing was performed using a notional KE slug with a mass of 8.800 kg
- This work built on previous efforts
 - DM63 Development by Rheinmetall
 - SCDB Propellant Evaluation by PM-MAS
 - GD-OTS IRAD
- This work involved every component of the propulsion system including:
 - Combustible Cartridge Case
 - Primer
 - Propellant



LEADING THE WAY

Slug Cartridge



LEADING THE WAY

Propulsion System Components



SCDB Granular Propellant



Stick Propellant

M123 Primer (used with stick propellant in M830/M831 rounds)



M35L70 Primer (used in this work)



LEADING THE WAY

Problems Encountered

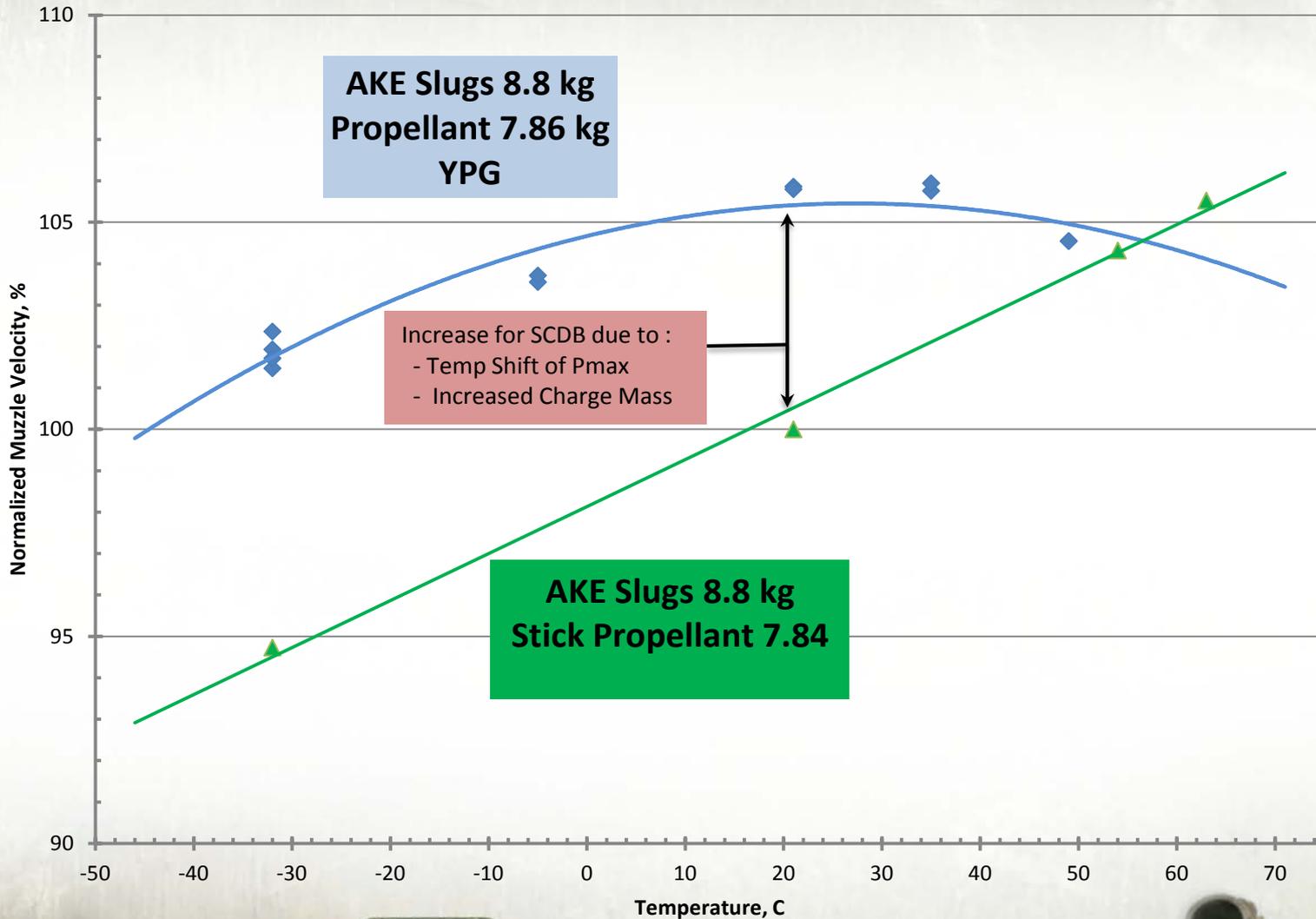
- SCDB Propellant Coating Strength
 - Dependent on loading density
 - Dependent on strength of ignition
- Ignition Issues due to the long aft boom intrusion and short primer
 - High Negative Delta Pressure
 - Flareback at cold temperature
- Ignition Issues solved by:
 - Changing primer configuration
 - Method of affixing propellant sticks
 - Integration of cartridge components



LEADING THE WAY

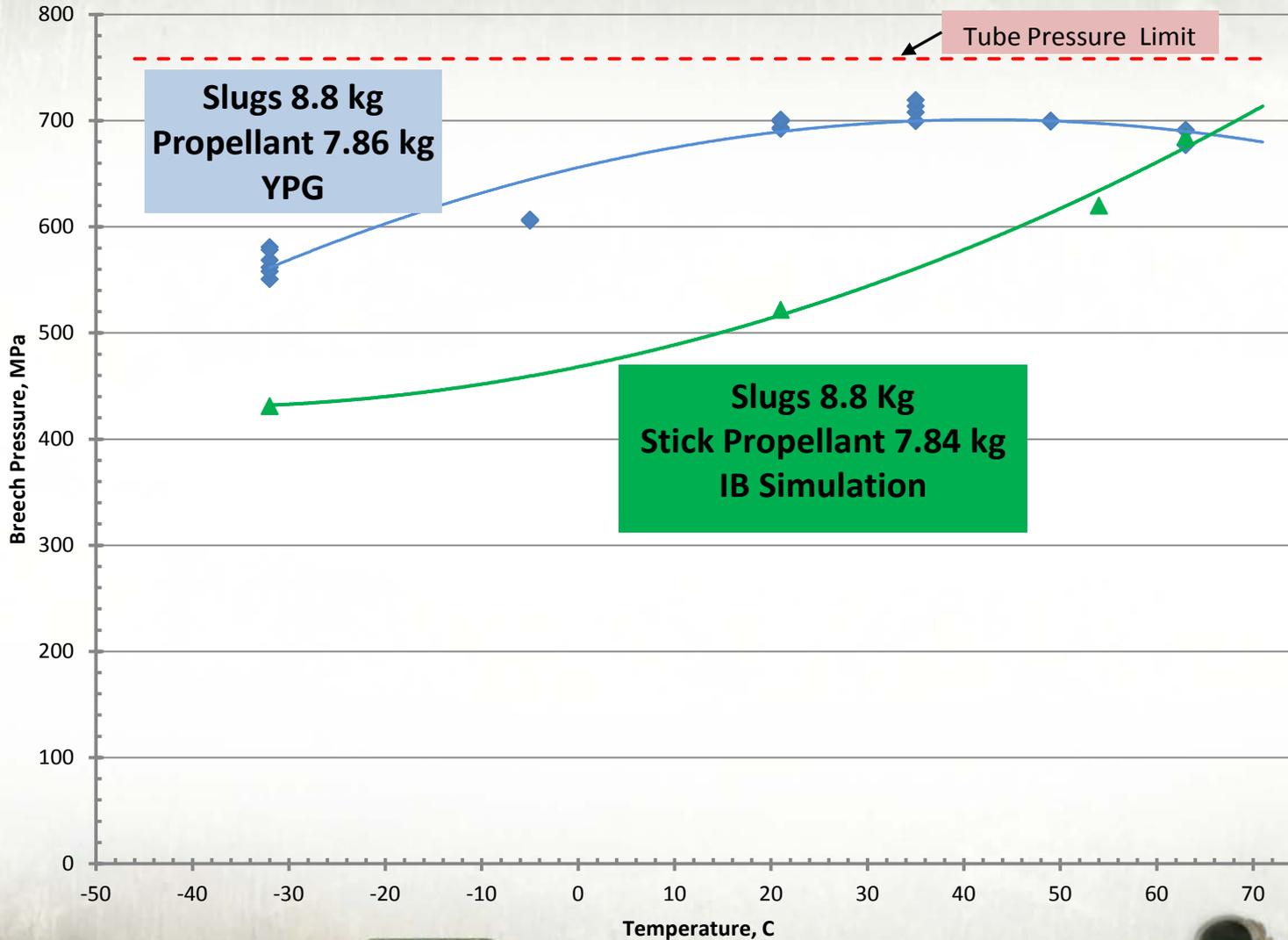


Final CV Test – Velocity Data



LEADING THE WAY

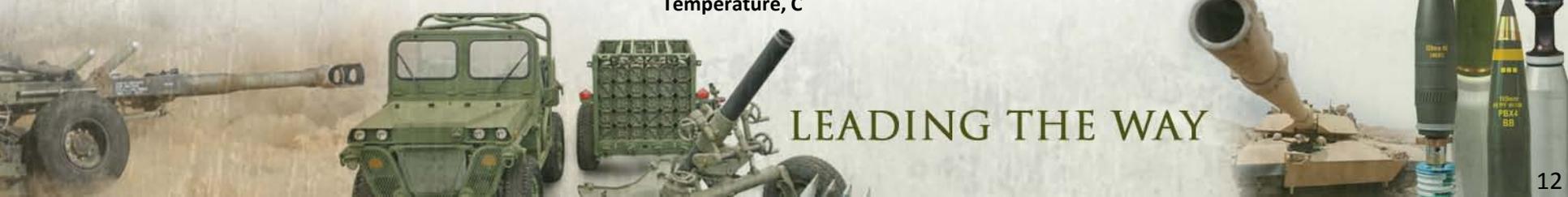
Final CV Test – Pressure Data



**Slugs 8.8 kg
Propellant 7.86 kg
YPG**

**Slugs 8.8 Kg
Stick Propellant 7.84 kg
IB Simulation**

Tube Pressure Limit



LEADING THE WAY

Summary

- This program successfully demonstrated a Thermally Insensitive Propulsion System (TIPS®) with a KE slug in 120mm Tank Gun application
- The outcome was demonstration of a propulsion system that only varied 65 m/s across temperature range from cold to hot
- Demonstrated a significant increase in velocity at all temperatures below 35°C
- All of this was achieved within the safety pressure limits of the current 120mm cannon



LEADING THE WAY