

"EDA Joint project on Insensitive Munitions & Ageing (IMA): overview"

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Background

→ Great advances in ammunition development, for the last 20 years

- Safety and Vulnerability are of major concern for ammunition
- IM concept has been developed
- Today IM are “In-Service” Munitions



→ And now...

- IM signature is assessed on pristine ammunition in ambient conditions
- Life duration extension context
- Multiple use, in different operations theatres

What about IM label, during ammunition life cycle ?

Ageing aspects versus IM assessment

→ **New concern emerging with IM introduction in forces**

→ **Adapted for a European cooperation, within the framework of EDA**

- Common European research topic
- Complementary approaches :
 - ✓ A great scope of applications, ammunitions
 - ✓ Wider range of methodologies : trials, modelling ...
 - ✓ Large database available



→ **IMA (Insensitive munitions and Ageing) : one of the first EDA project, under French leadership.**

I. IMA project overview

II. Ageing programs and first results

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IMA project overview

Overall objectives

→ Need

- To quantify and predict if ageing can influence ammunition vulnerability
 - ✓ To take into account ageing aspects and their consequences on safety and vulnerability aspects, from the early development steps
 - ✓ To have a dedicated and reliable methodology for vulnerability assessment, that would take into account ammunition real life cycle



→ Final goal

- To be able to predict ammunition IM signature, for any ammunition all along its life cycle



IMA project objectives

→ IMA project : a study focused on Energetic Material

- **To investigate ageing effects on EM vulnerability characteristics**, considering factors such as:
 - vulnerability behaviour (insensitiveness properties changes ...),
 - mechanical properties,
 - chemical properties...
- To study and develop **surveillance methods** to characterize materials properties with ageing
- And then to offer **a first step of an overall methodology** able to predict if an IM signature assessed on pristine ammunition can be kept all along its life cycle



Project organization

→ 7 participating nations (MoD & Industries)

- Czech Republic : Explosia
- Germany : ICT / ISL
- Finland : PvTT
- France : Herakles (former SNPE)
- the Netherlands : TNO
- United Kingdom : DSTL
- Sweden : FMV



→ 4 technical "Work Packages" (WP), described in a Technical Arrangement

- State-of-Art (WP1)
- **Predictive methodologies** : identification of major safety / vulnerability parameters and characterization through ageing (WP2)
- Selection of pertinent parameters for **surveillance**, with dedicated methodologies (WP3)
- Synthesis (WP4)

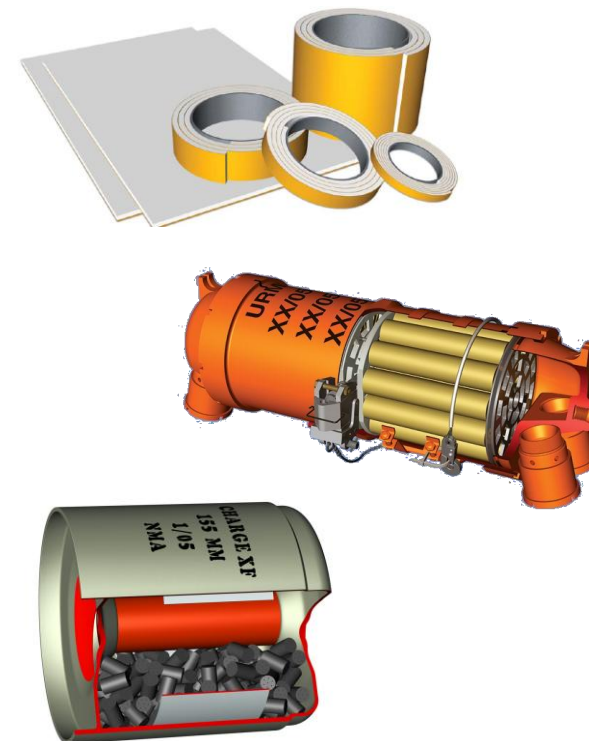
→ Project total duration : 5 years, started in January 2009

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Ageing programs and first results

→ Compositions

Type	Product name	Composition
High Explosive	Semtex® PI SE	M1RDX/polymere (88/12)
	Semtex® PI SE	HMX/polymere (88/12)
Rocket propellant (double base propellant)	ROP	NC/NG/Cl/DNT
	ROP1	NC/NG/Cl/substitution of DNT
Gun propellant (triple base propellant)	D380	NC/NG/NQ/Cl



→ Test matrix definition

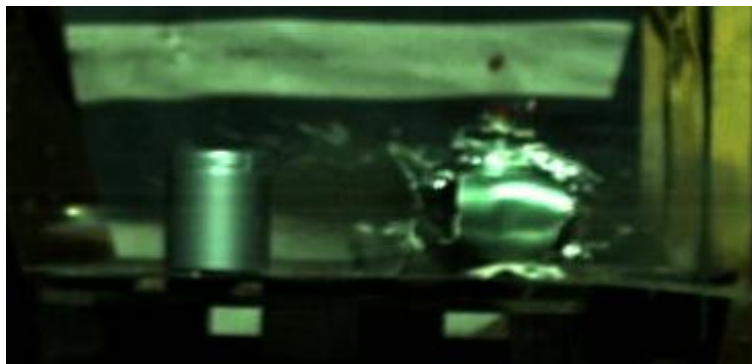
- **Quantitative analysis** - HPLC/UV, GC/MS
- **Chemical stability tests** - Test @ 100°C (HT100), Bergman Junk test at 120°C, Methyl - violet test
- **Sensitivity & stability tests** – TAM, HFC, VST - Vacuum stability test, DSC, DTA
- **Mechanical tests** - Friction sensitivity test, *BAM fall-hammer* Impact sensitivity test
- **Impact & shock sensitivity tests** - BI test, SCJI test

→ Ageing conditions

	⇔ 10 years, 25°C For EM without NC		⇔ 10 years, 25°C For NC based EM	
Temperature of ageing (°C)	[days of ageing]		[days of ageing]	
80	24		10.6	
89	10.5		3.83	
For NC based propellant	⇔ 5 years	⇔ 10 years	⇔ 15 years	⇔ 20 years
65	32,4	64,9	97,3	129,7

→ First results / comments

- No evolution after ageing (80 / 89°C) is observed through BI tests



Modular charge

Bullet impact
test @ 12.7mm
**B-32 API &
BZT API-T**



Rescue rocket motor

→ Compositions

- Two PBX formulations are included in Finnish IMA work, FOXIT and FPX 7
 - Both sample explosives include the same ingredients but having different sensitivity properties
 - The basic composition of the samples are HTPB / RDX / AP / Al and both are cured with IPDI

→ Test matrix definition

- Several tests were conducted to find out the effect of ageing
- Testing program included testing of sensitivity, thermal, mechanical properties as well as chemical analysis

Sensitivity / GAP tests	Thermal properties	Mechanical properties	Chemical analysis
EIDS GAP (FOXIT)	Deflagration point	Tensile test (+23 °C & -40 °C)	FTIR (double bonds)
LSGT (FPX 7)	Heat Flow Calorimetry	Dynamical Mechanical Analysis	HPLC (antioxidant)
/	Differential Scanning Calorimetry	/	/

→ Ageing conditions

- Artificial Ageing program is based on Finnish long term storage conditions
 - Ageing is conducted at different temperatures
 - Ageing times at each temperature were selected so that they represent storage for 5, 10 and 15 years
 - Ageing times were calculated according to van't Hoff rule

Corresponding storage time (years)	Ageing Temperature and ageing times in days				
	40 °C	50 °C (f = 2,0)	50 °C (f = 2,5)	60 °C	70 °C
5	154	140	62	25	10
10	308	281	123	49	20
15	400	365	160	64	25

→ First results / comments

- Test program almost finished, results to be further analyzed

→ Compositions

- 2 PBX : PBXN109 (HTPB / Al / I-RDX) and B2214B (HTPB / NTO / HMX) – EURENCO’s PBX
- 1 classic HTPB / AP/ Al propellant (reduced smoke)

→ Test matrix definition

- Focus on SH / BI / SR threats
- Parameters to be evaluated through ageing:

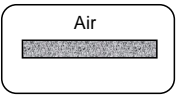
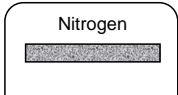
EM \ Threat	Mechanical	Slow Heating (SH)	Bullet Impact (BI)	Sympathetic Reaction (SR)
Rocket propellant	Uniaxial tensile test, DMA	DTA	Friction sensitivity	/
		SCO test	Friability (Shot gun), High pressure Combustion	/
PBXN109	Uniaxial tensile test, DMA	DTA	Friction sensitivity	/
		SCO test	/	/
B2214B	Uniaxial tensile test, DMA	DTA	Friction sensitivity	Micro GAP test
		/	Friability (Shot gun), High pressure Combustion	/

Sensitivity

Violence of Reaction

- Validation tests on mock up (BI / FI / SCO), on both non aged and aged (equiv. 20 years) EM

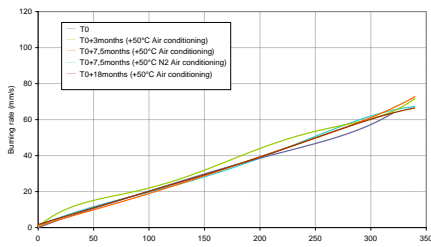
→ Ageing conditions

Position	Ageing T° RH dry	Ageing time (Years)	Ageing time (Months)	Corresponding storage time (years)
REFERENCE T0		0	0	0
Surface 	+20°C	2	24	2
		5	60	5
		10	120	10
	+50°C	0,25	3	2
		0,625	7,5	5
		1,5	18	12
In depth 	+20°C	5	60	5
		10	120	10
	+50°C	0,625	7,5	5
		1,5	18	12
		2,5	30	20
		3,125	37,5	25

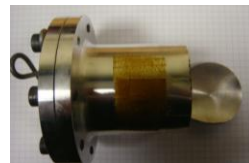


→ First results / comments

- Hardening of EM is observed, especially on rocket propellant
- No significant evolution on safety / vulnerability tests, after 20 years equiv. ageing
- 2 years @ +20°C ⇔ 3,5 months @ +50°C

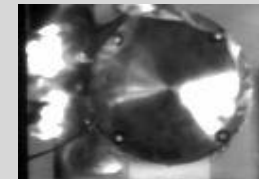
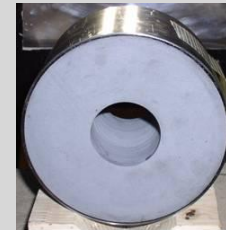


B2214B rb evolution



SCO test on PBXN109

Rocket propellant mock up



~ 30% of unburned propellant is recovered

Fragment Impact @ 1888 m/s → Type III-IV reaction
(Reference test)

→ Compositions

- 4 compositions of high explosive charges (HEC) are made and investigated by Germany

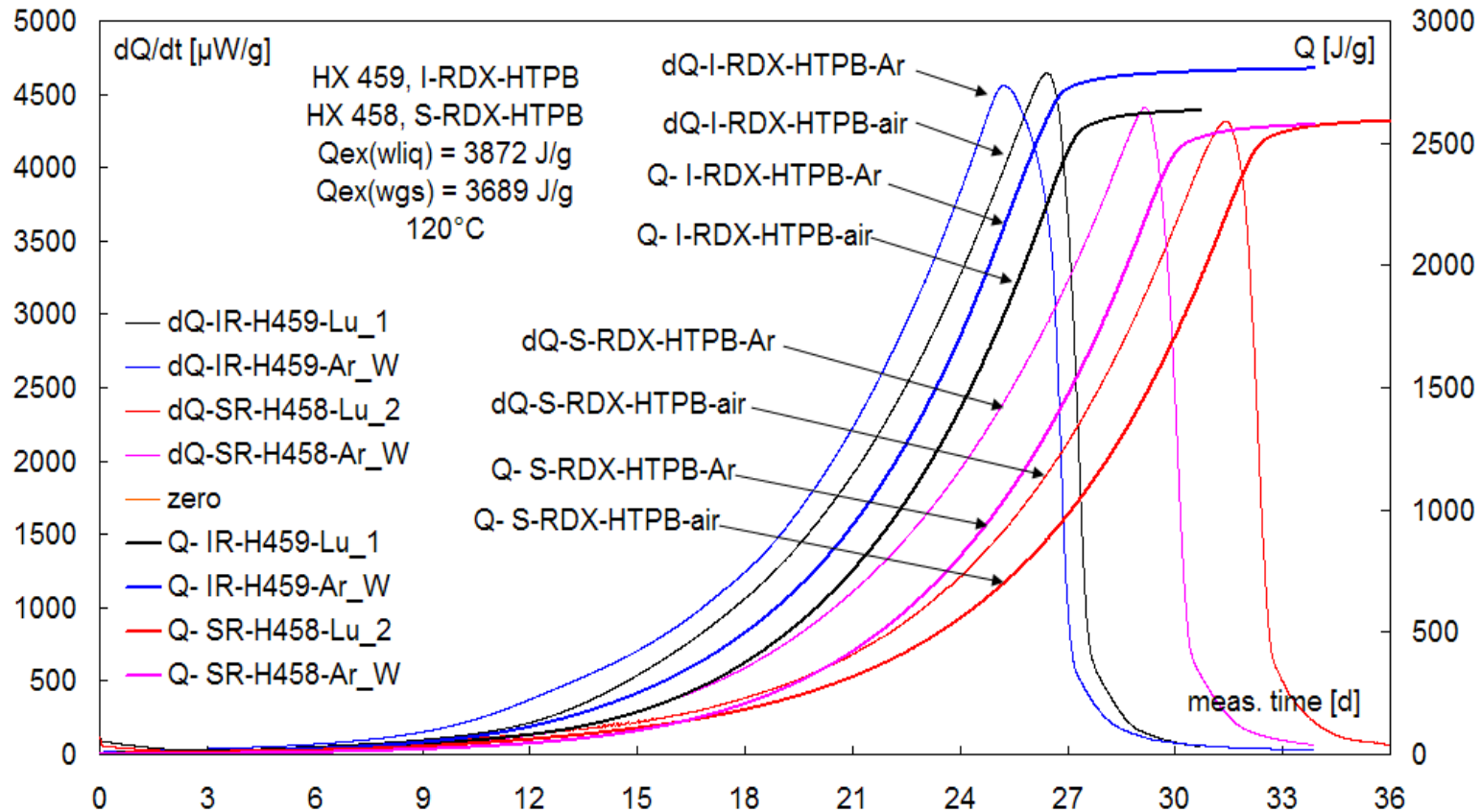
<p>Used components</p> <p>I-RDX: insensitive RDX from EURENCO S-RDX: standard RDX from Dyno</p>	<p>GAP diol, I-RDX / S-RDX, class 1 and 5 HTPB R45M, I-RDX / S-RDX, class 1 and 5</p>
<p>Composition (high explosive charge, HEC)</p> <p>Two formulations with I-RDX (class 1 and 5)</p> <p>Two formulations with S-RDX (class 1 and 5) - for comparison to I-RDX formulations</p>	<p>GAP-N100 bonded I-RDX / S-RDX</p> <p>75% RDX (I or S), 15% binder GAP-N100, 10% plasticizer BDNPA_F</p> <p>HTPB-IPDI bonded I-RDX / S-RDX</p> <p>80% RDX (I or S), 12% binder HTPB-IPDI (+AO), 8% plasticizer DOA</p>

→ Test matrix definition

- Basic characterization of energetic components (I-RDX) and HEC :
Impact and friction sensitivity / Vacuum stability / Auto ignition temperature / Density
- Tests :
 - ARC (Accelerating Rate Calorimetry)
 - DSC (Differential Scanning Calorimetry)
 - HFMC (Heat Flow Micro-Calorimetry)
 - DMA (Dynamic Mechanical Analysis)
 - 21 mm GAP detonation sensitivity
 - 30 mm bullet (12.7mm) impact test

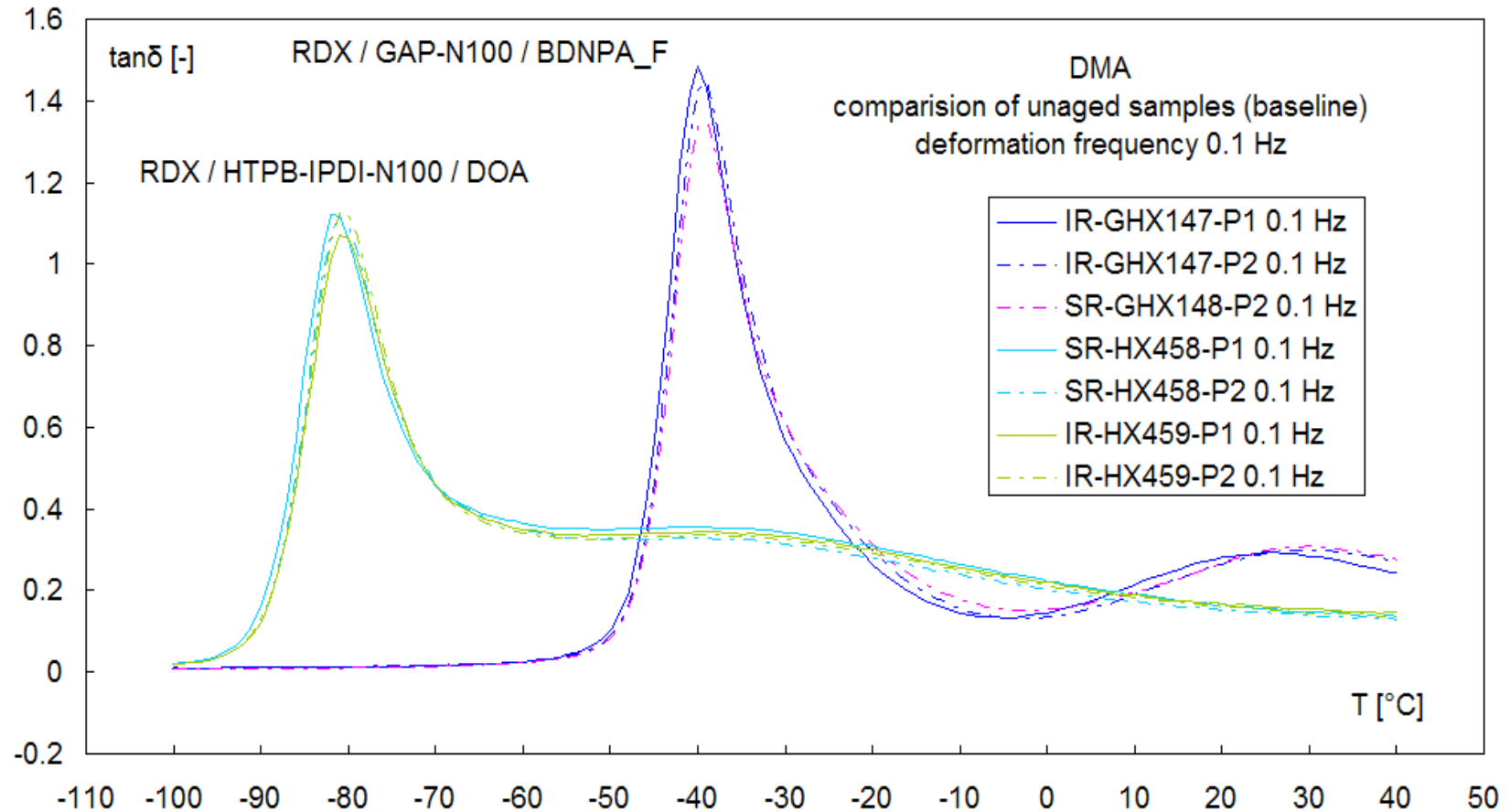
→ First results / comments

- Microcalorimetry: Heat Generation Rate dQ/dt and Heat Generation Q at 120°C, in air and in argon; comparison of I-RDX-HI and S-RDX-HI formulations



→ First results / comments

- Comparison of loss factor of I-RDX and S-RDX HI and GN formulations (≠ bonded)

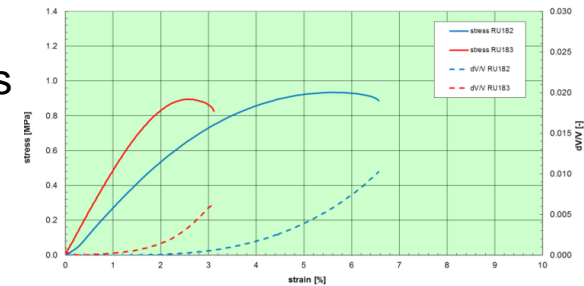


→ Compositions

- 2 Bimodal RDX-HTPB-based explosive (85 wt% solid loading) with NCO-OH ratio of 1.2 and 1.4 (last one is “chemically aged”), first one artificially (high T) aged

→ Test matrix definition

- IM testing : Fragment impact and Cook-off (0.06°C/s and 0.006°C/s)
- Friability testing
- Gas DilatoMetry (GDM tensile strength test with $\Delta V/V$) →
- SEM...



→ Ageing conditions

- Ageing at 70°C in contact with air inside the oven

→ First results / comments

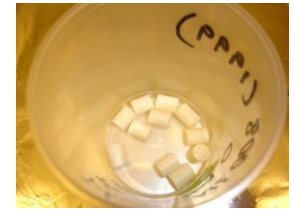
- Detonation in reference testing of SCB at 0.06°C/s
→ more investigation of the cause of this unexpected event
- In middle of test series and analyses



→ Compositions

- 2 LOVA-propellants NL007 and NL008 have been tested.
- 1 propellant lot of NL008 without graphitization has also been analyzed

Component	NL007	NL008
RDX	76.5 %	73 %
CAB (cellulose acetate butyrate)	9 %	12.7 %
NC (nitrocellulose)	7.2 %	5.3 %
TBC (Tributyl citrate, plasticizer)	7 %	8.6 %
Centralite I (NC-stabilizer)	0.4 %	0.4 %



→ Test matrix definition

Analysis method	Property studied
Uniaxial Compressive Testing	Mechanical properties
Density	
Vented bomb	Grain capacity to withstand high pressure. Burning properties in general.
Slow cook-off	Thermal properties
DSC	
HPLC	NC-stabilizer content

→ Ageing conditions

- Ageing temperatures: 50°C, 65°C and 80°C.
- Storage temperature: 25°C
- Sample collection at 0, 1, 5, 10 and 20 years of forced ageing.
- Thermal cycling

K_T = Thermal acceleration factor
 T_1 = Storage temperature
 T_2 = Ageing temperature
 E_a = Activation energy

Simulated age = K_T x storage time at elevated temperature →

$$K_T = e^{\frac{E_a}{R} \left(\frac{1}{T_1} - \frac{1}{T_2} \right)}$$

→ First results / comments

- No severe negative impact of ageing / thermal cycling on properties for graphitized propellants.
- Probable splitting of grains at combustion of aged non-graphitized propellants.
- NL007 : ≠ trends in results between ageing @ 50°C and ageing @ 65°C & 80°C.
- Uniaxial compressive testing does not seem to be a suitable test method for studying the ageing behaviour of these propellants due to either propellant composition or geometry.
- Vented bomb analysis seems to be a promising method for propellant surveillance.

→ Compositions

- Torpedo warhead PBX (AP/ RDX / AI / HTPB)

→ Test matrix definition

- Tube tests :
 - Fragment impact
 - Fast Heating



→ Ageing conditions

- 6 months at 80°C (<10% relative humidity)

→ First results / comments

- Test program to begin FY 2012-13 for contractual reasons

Conclusions

→ IMA project status, after 3 years of work :

- 7 European nations are participating to the project
- Ageing programs are going on, involving different EM, experimental protocols...
- Numerous data are now been gathered, and need to be further analyzed

→ IMA project :

- a first step of an methodology → predicting if an IM label is still valid after ageing

→ IMA project : a successful EDA cooperation

→ To be followed...

Acknowledgment



And all participating nations MoDs.

KEY MISSIONS, KEY TECHNOLOGIES, KEY TALENTS