

TECHNOLOGY DRIVEN. WARFIGHTER FOCUSED.

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Distribution A: Unlimited Distribution



### **Presentation Structure**



A. Program Overview & Update

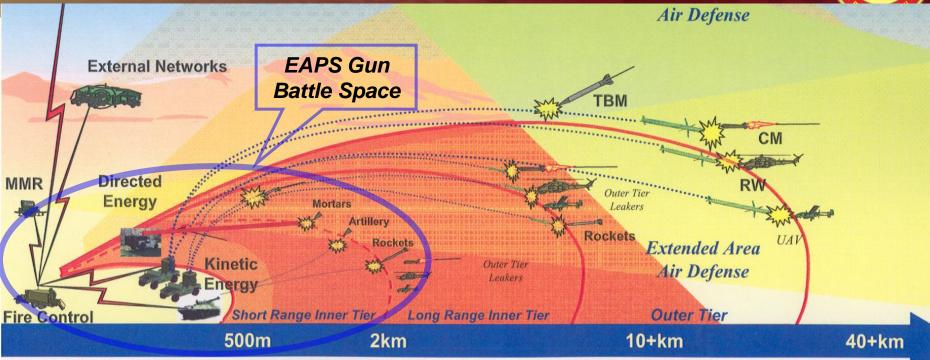
B. Flight Body Evolution of EAPS Interceptor





### **Program Goal and EAPS Battle Space**





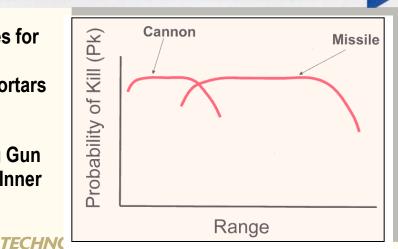


#### **Target List**

- Rockets: 107 mm-240 mmArtillery: 122 mm-152 mm
- Mortars: 60 mm 120 mm

- Goal is to Develop Technologies for 360 Degree Mobile Air Defense Against Rockets, Artillery and Mortars (RAM)
- ARDEC ATO Program Pursuing Gun Based Solution for Short Range Inner Tier – Need to fill range Gap of

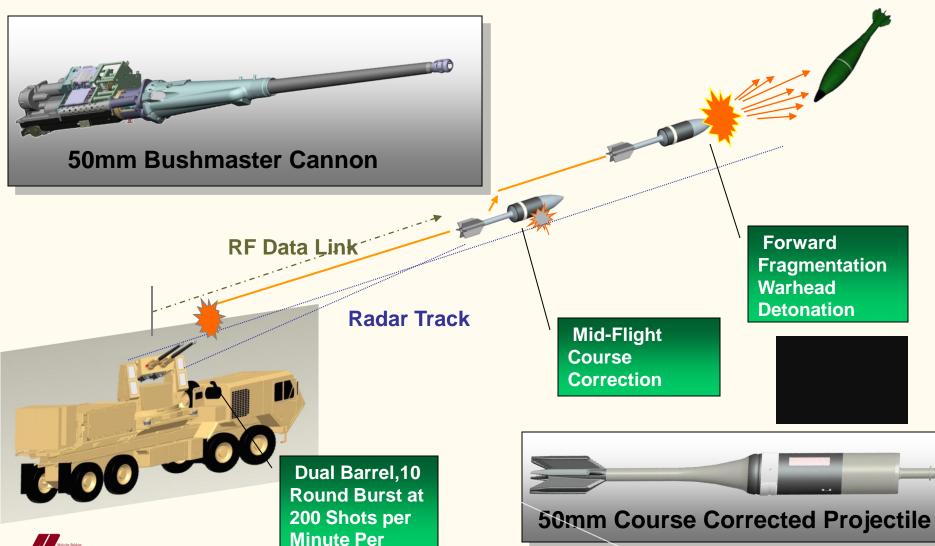
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## EAPS Gun Baseline Concept







# Phase II Program Goals and Objectives



- Demonstrate a prototype 50mm auto cannon that fires at 200 rds/min w/ a feed system for a ten round burst.
- Demonstrate a fire control sensor & commo station to simultaneously track burst of ten interceptors and two threats and command the ten interceptors to maneuver and warhead function.
- Demonstrate a 50mm cartridge to meet threshold performance.

Demonstrate integrated system (System Level TRL-6) by defeating two stationary threats in a simultaneous emulated scenario.





# EAPS Gun ATO Integrated Demo Roadmap



Technology
Demonstration
Plan

Baseline Gun (112rds/min)
Baseline Projectile Design
New E-Scan Radar

Fixed Gun,
Stationary Targets
Demo

Indiv

Demo

Component

Performance Demonstration Plan Revisit Syst Study For Optimum Caliber High Rate Gun (200rds/min)
Optimized Projectile Design
New E-Scan Radar

Smaller Battery
MEMS S&A
Reduced Antennae
Smaller Electronics
More Lethal Warhead





## EAPS Gun System HEMTT Mount





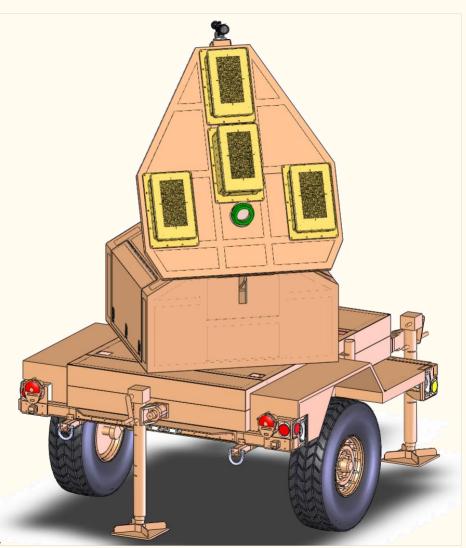
- C5, C17, C130 Transportable
- Meets Mobility Requirements
- More Cost Effective Than the Stryker Stryker Also An Option





# Technovative Applications EAPS Fire Control System





512-element Transmit Antenna based on PPS with improved cooling

512-element Receive Antennas (3) based on PPS

1 meter interferometer baseline for high angular accuracy

PPS electronics group with upgraded processors

Fire control software adapted from PPS

Multi-target tracking and RF communications







# FLIGHT DESIGN REFINEMENT TO REDUCE AEROBALLISTIC DRAG



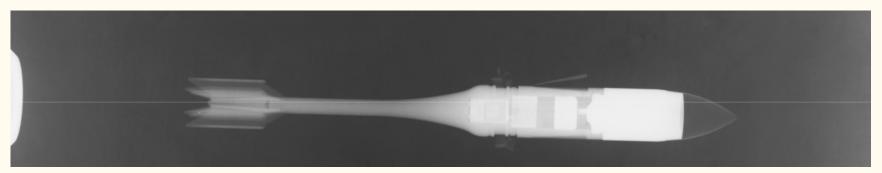


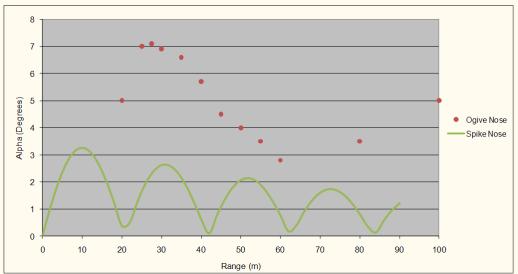
## Original Ogive Design





Flight Design, Nov 2007









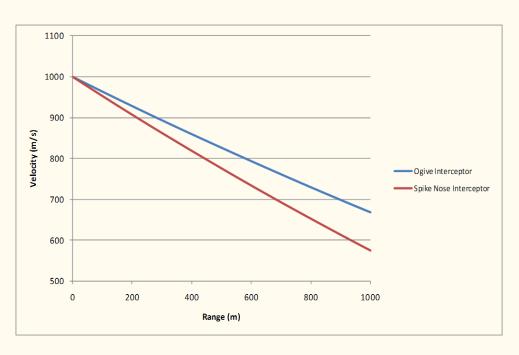
### Spike Nose Fix - Demo Round



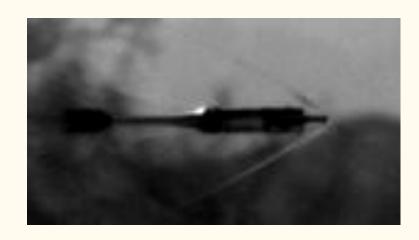


Special Specia

Flight Design from Feb 2008 to Mar 2009, Demo



**Velocity Comparison** 



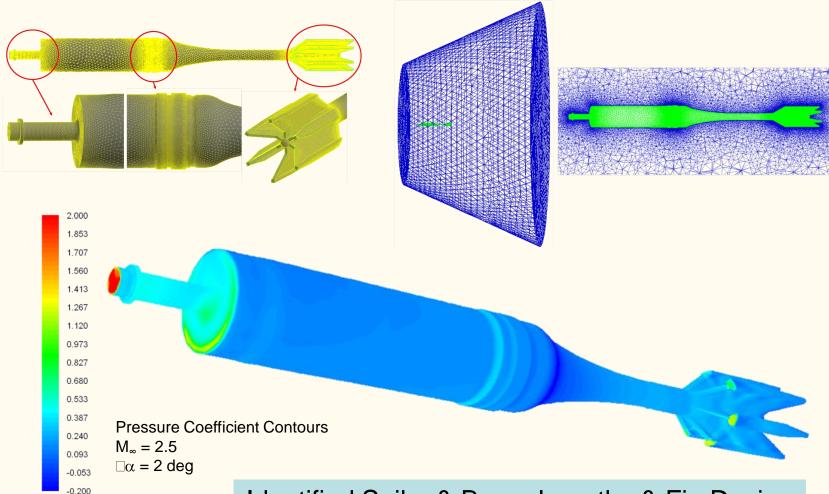
Increased Stability at the Expense Of Drag





## **Computational Fluid Dynamics**







Identified Spike & Boom Lengths & Fin Design As Variables for Drag Reduction

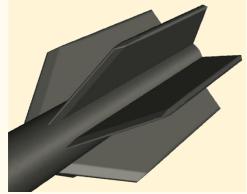


### **Test 405 Matrix**





1.5 Caliber Spike



No T-Tab Fin



1.5 Caliber Shorter Boom, T-Tab Fin



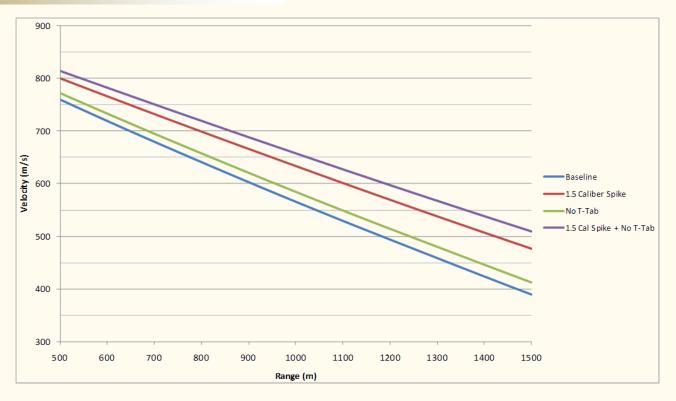
### 1.5 Caliber Spike, No T-Tab Fin

Interceptor description	@ Mach 2.5	
	Prediction	Test 405
1.5 caliber spike nose	0.57	0.54
No T-Tab fin	0.61	0.63
_ 1.5 caliber spike + no T-Tab fin	0.54	0.50



# RDECOM Drag Reduction Summary





	Velocity @ 1000 m	Difference in velocity from test 303	
	[m/s]	[m/s]	
Test 303	567	-	
1.5 caliber spike	634	+67	
No T-Tab fin	585	+18	
1.5 caliber spike	658	+91	
+ no T-Tab fin	038	<del></del>	





### Summary



Fire Control Sensor Contractor Allows Program to Move Toward Integration Phase

Demonstration in FY14.

Interceptor Optimization (still ongoing) Will Meet Performance Requirements

Lethality Assessment tests to optimize warhead design.

System study to select optimum caliber.

Flight design and refinement to reduce aeroballistic drag.

