Disruptive Technology: Hypersonic Propulsion

September 2007



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Cleared for Public Release: # AFRL HQ 07-0083







• The value of speed

• What does it take

• Summary



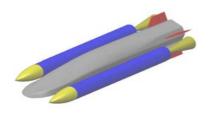
Potential Air Breathing Hypersonic Applications



Source: SAB, Why and Whither of Hypersonic Research in USAF, Dec 2000

Weapons (missiles)

- Time Critical Mobile targets
- Hard and Deeply Buried Targets
- Suppression of Enemy Air Defense
- Ballistic Missile Defense / Theater Ballistic Missile Defense



SAB: Scientific Advisory Board SAB TR 00-03, Cleared for Open Distribution

Aircraft

Global Strike/Recce



Space Operations

- Routine launch
- Replace & maintain key satellites

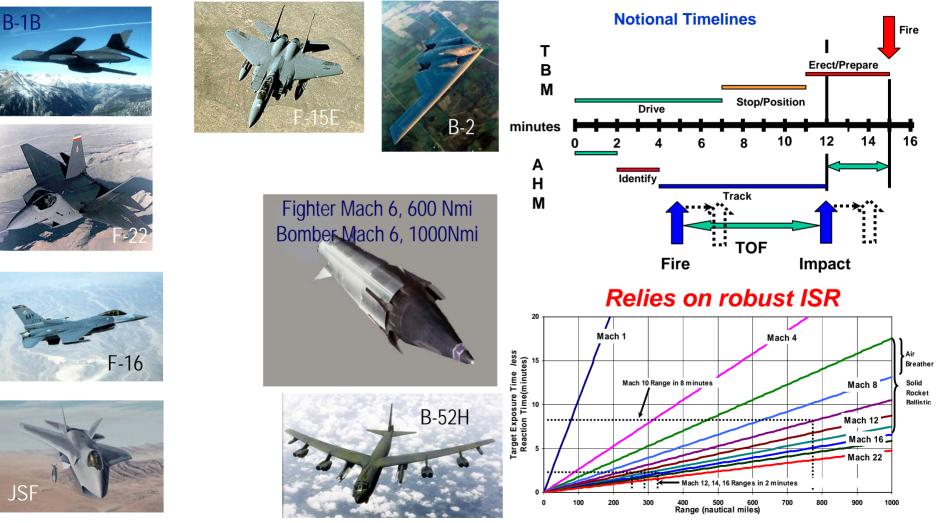




Air Breathing Hypersonic Missile



Source: SAB, Why and Whither of Hypersonic Research in USAF, Dec 2000

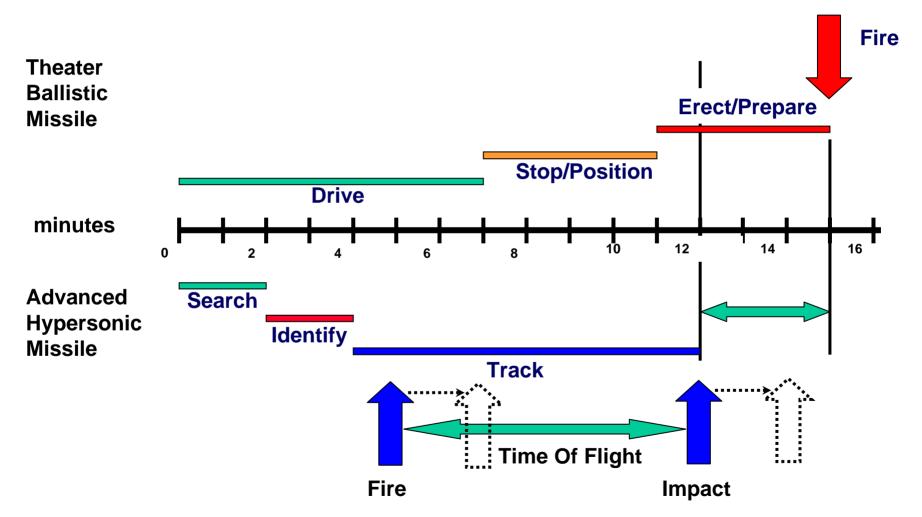


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Notional Timelines

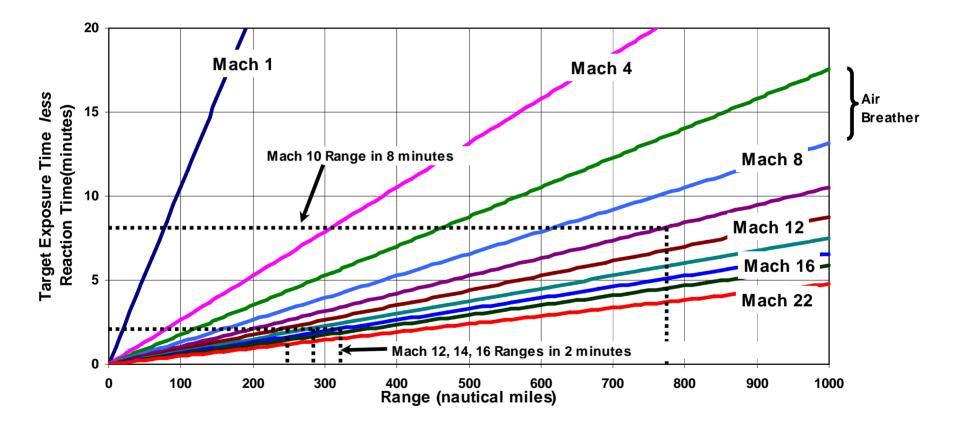
Source: SAB, Why and Whither of Hypersonic Research in USAF, Dec 2000



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Aircraft: Mach 5-7 Strike/Recce Vehicles



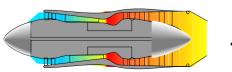


Pros:

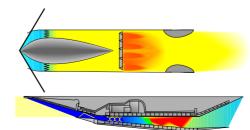
- Limited infrastructure impact
- Very fast response times
- Increased speed helps survivability
- Storable fuels capable of standing alert

Cons:

- High production costs
- Some R&D needed: combined cycle engines, materials, and cooling
- Increased cost for maintenance, logistics



Cleared for public release: ASC 03-2959



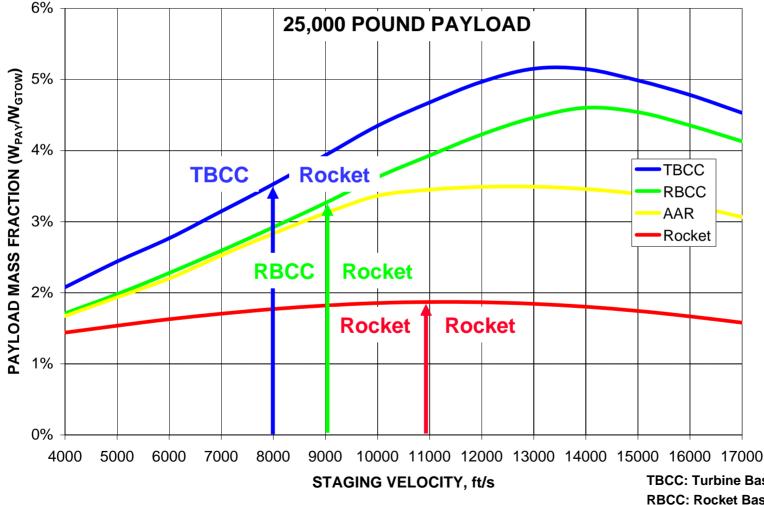
Typically Turbojet and Ramjet or Scramjet



Mach 8 Air-Breathing Propulsion Doubles Payload Fraction to LEO



PAYLOAD MASS FRACTION vs STAGING VELOCITY FOR H₂ - O₂ TSTO VEHICLES TO 51.7° ORBIT

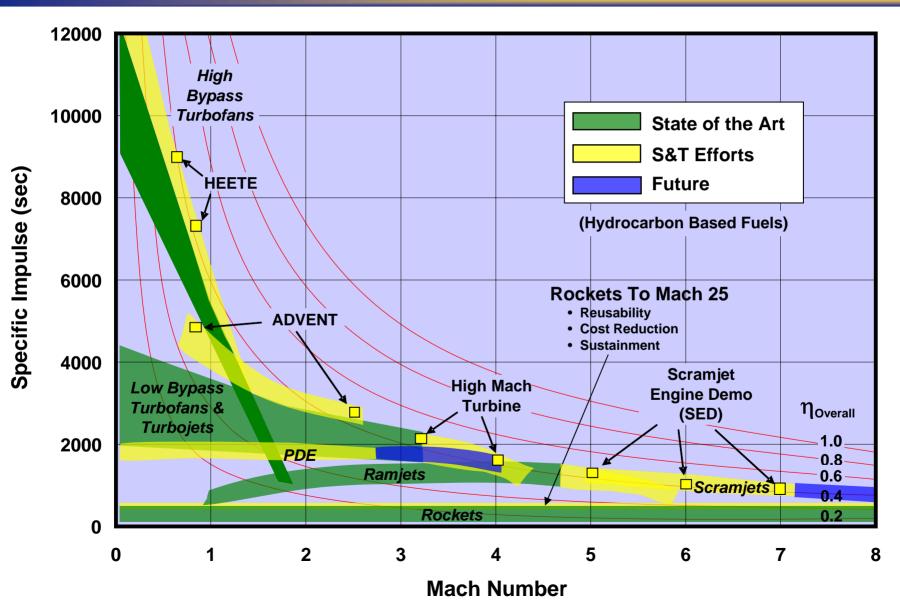


Cleared for public release: ASC 03-3191

TBCC: Turbine Based Combined Cycle RBCC: Rocket Based Combined Cycle AAR: Air Augmented Rocket



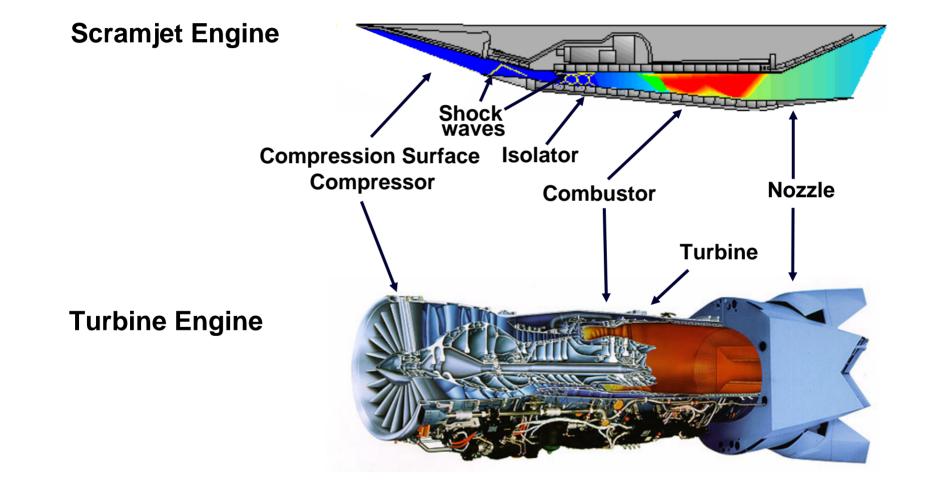
Aerospace Propulsion Domain Improving Fuel Efficiency Across Flight Spectrum



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Scramjet Propulsion

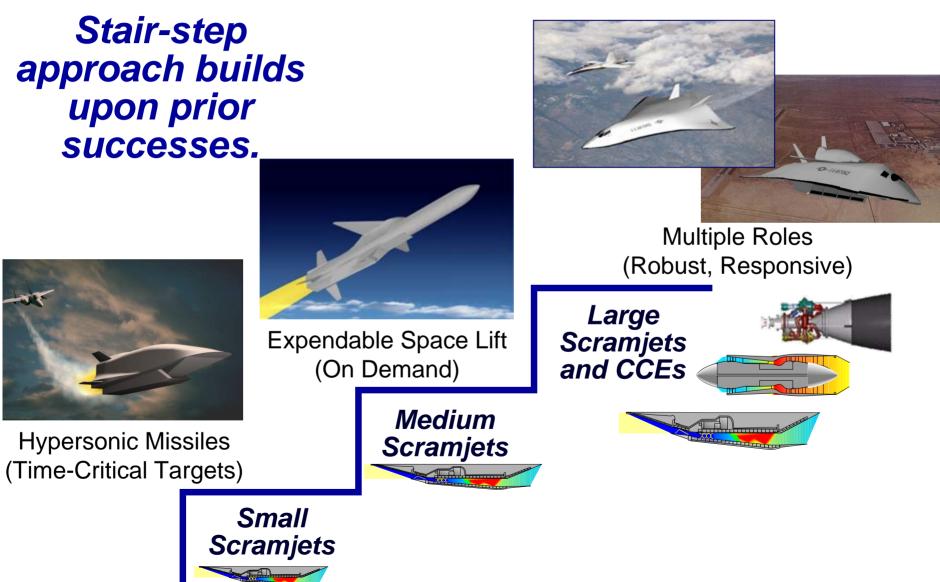






AFRL's Hypersonic Technology Development Approach

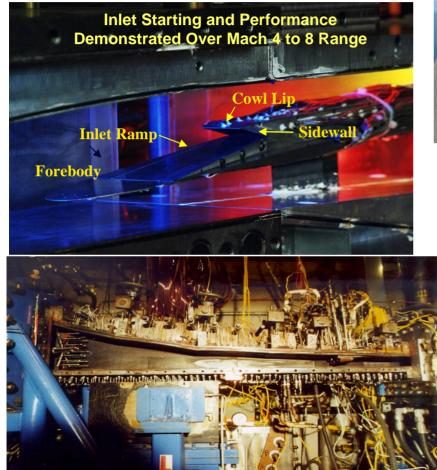






HyTech: HYDROCARBON SCRAMJET COMPONENT TECHNICAL CHALLENGES HAVE BEEN MET



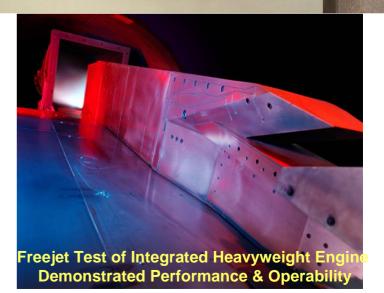


Combustor Operability and Performance Demonstrated Over Mach 4 to Mach 8 Range

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6"X30" & Full 75" Sidewall Fuel-Cooled Heat Exchanger Demonstrated Structural Durability



Ground Demonstration Engine 1 Status: Testing at GASL Completed





- Testing initiated Sep 02, completed Jun 03, 60 tests
- Mach 4.5: Performance comparable to heavyweight PTE tests
- Mach 6.5: Performance exceeded heavyweight PTE tests
- Engine fuel cooled structure in excellent condition

First Demo of Flight Weight Hydrocarbon-Fueled/Cooled Scramjet Engine!



Ground Testing Carries Scramjet Development Only So Far



High-speed flight simulation on the ground is tough

- Limited facilities world-wide
- Even in the very best facilities compromises are made
 - Model size and structure
 - Inlet air purity ... composition, T-P-M profiles, etc
- Testing at fixed Mach number ... no flight transients
- Still have to deduce installed performance ... for an engine-airframe that must be well integrated
- Hydrocarbon fuels and active structural cooling exacerbate the test challenges
 - Complex fuel thermodynamics and chemistry

Flight-test required to demonstrate scramjet viability



X-51A Scramjet Engine Demonstrator (SED)





- Joint AFRL-DARPA flight demo of AFRL's hydrocarbon-fueled scramjet & waverider airframe technologies
- Uses modified ATACMS booster
- Scramjet take-over at Mach 4.5
- Cruise at Mach 6.5 to 7.0
- Four flights (FY09 1st flight)

X-51A SED Configuration

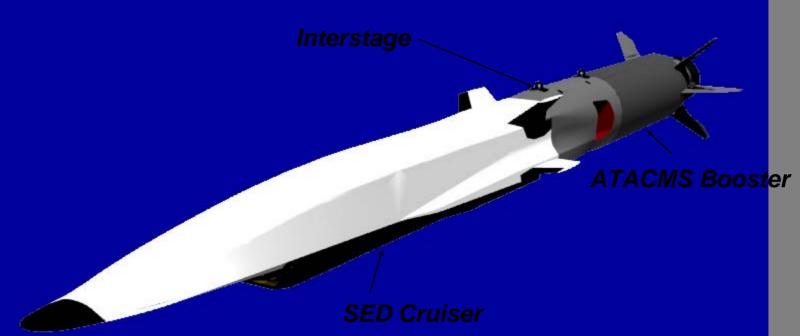


Cleared for Public Release: # DOD 06-S-2154

Potential Weapon Characteristics

- Tandem or side-by-side booster
- 2300 lb launch weight
- Range: 600 nm/10 min
- 300 lb payload (penetrator, smart submunitions, or explosive)





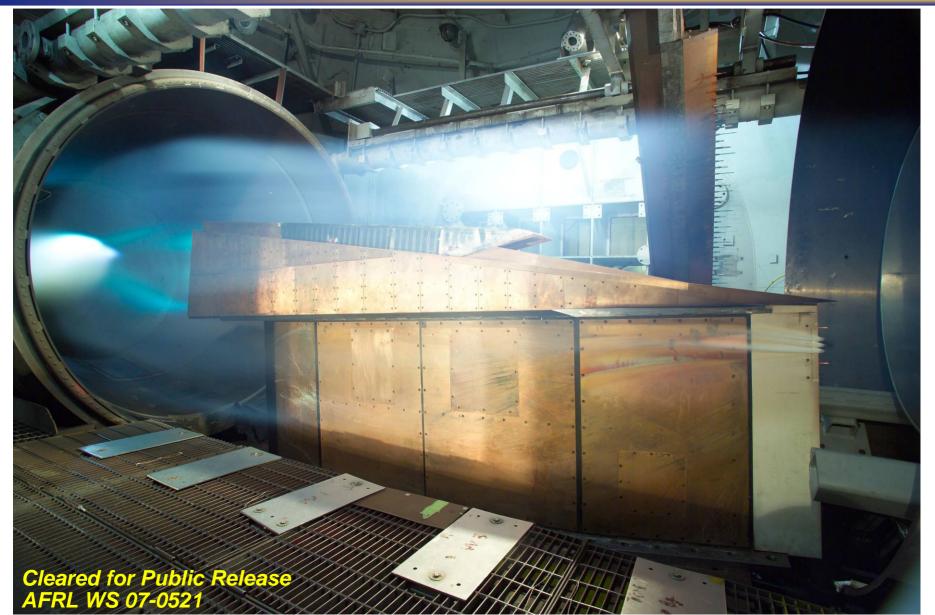
Utilizes one HyTech scramjet engine derivative flowpath **Composite leading edges and Boeing proprietary TPS**

 Utilizes off-the-shelf booster technology; booster will be modified slightly to optimize air launch 16



SJX61-1 Testing at LaRC

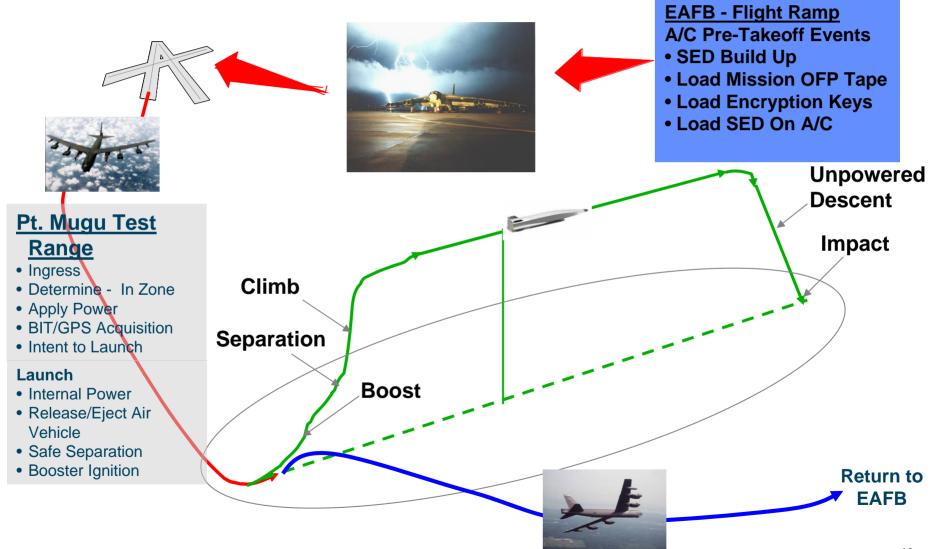


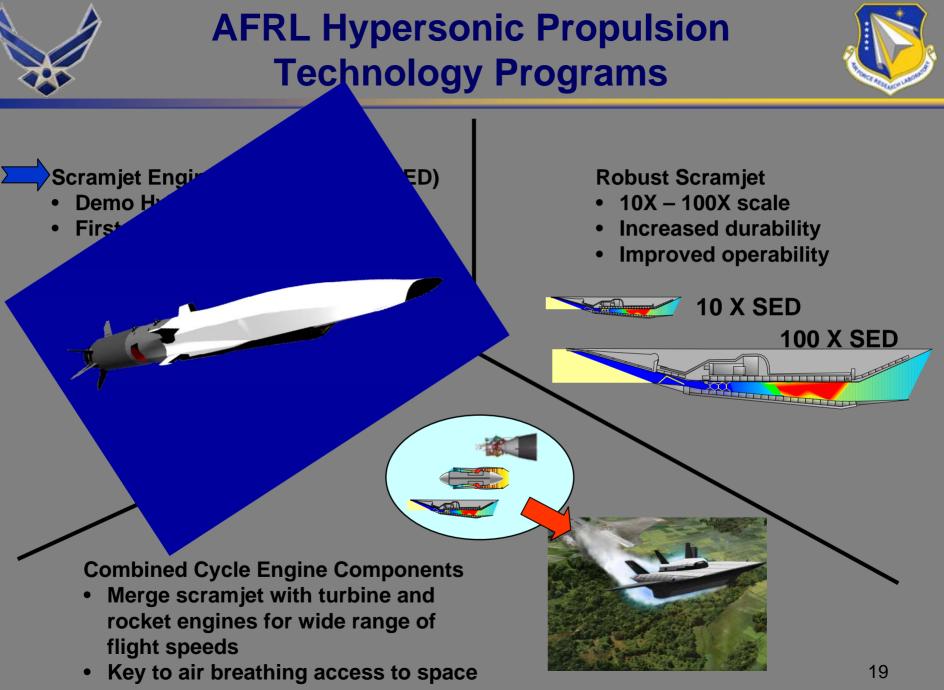




X-51 SED Flight Test Operations



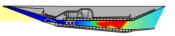


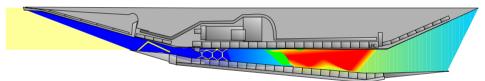


Robust Scramjet Program



- Addresses scaling issues for 10X and 100X flowpaths
 - Examines improvements to fuel penetration and mixing, as well as general combustion kinetics
- Improves overall scramjet
 operability
 - Reduce takeover to Mach 3.5
 - Extend upper Mach to 8 to 12
- Identifies new structural concepts to extend durability and reusability
 - 50 cycles near-term
 - 250 cycles far-term





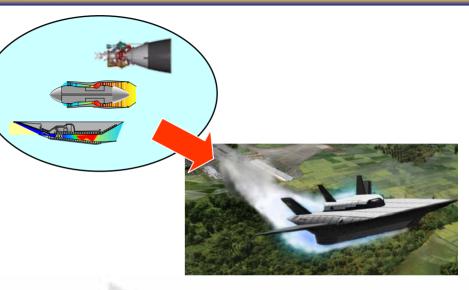




Combined-Cycle Engine Component Development



- Studies since 2002 show concept feasibility and no show stoppers
- Combines ramjet/scramjet engines with turbine and rocket engine cycles
- Rocket-Based Combined Cycle (RBCC) easiest integration
- Studies show promising results for over-under arrangements Turbine-Based Combined-Cycle (TBCC)
- Component development to begin October 2005









- Hypersonics is a disruptive technology
 - Near term: hypersonic cruise missile could be used to defeat time critical targets
 - Far term: CCEs could enable operationally responsive spacelift with "aircraft-like" operations
- AFRL's hypersonic propulsion technology development is on track to mature the enabling technologies



Propulsion is the Pacing Technology in Aviation



Wright Brothers Redesigned Engine Provided Enabling Powerto-Weight Ratio



Jet Engine Revolutionized the Shape and Speed of Aerospace Vehicles



High Bypass Turbofans Enabled Jumbo Jets



Scramjets Will Enable Sustained Hypersonic Flight and Routine Access to Space

